

Solid-liquid Flow in Stirred Tanks: Euler-Euler / RANS Modeling

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2 Solid-liquid Flow in Stirred Tanks: Euler-Euler / RANS Modeling
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12 **Abstract**
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15 Stirred tanks are widely used equipment in process engineering. CFD simulations of such equipment
16 on industrial scales are feasible within the Euler-Euler / RANS approach. In this approach,
17 phenomena on particle scale are not resolved and, accordingly, suitable closure models are required.
18 The present work applies a set of closure relations that originates from a comprehensive review of
19 existing results. Focus is on the modeling of interfacial forces which include drag, lift, turbulent
20 dispersion, and virtual mass. Specifically, new models for the drag and lift forces are considered
21 based on the best currently available description. To validate the model a comprehensive set of
22 experimental data including solid velocity and volume fraction as well as liquid velocity and
23 turbulence has been assembled. The currently proposed model compares reasonably well with this
24 dataset and shows generally better prediction compared with other model variants that originate
25 from different combinations of force correlations.

26
27 **Keywords:** stirred tanks, solid-liquid flow, Euler-Euler two-fluid model, closure relations,
28 Reynolds-stress turbulence model
29

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30 1 INTRODUCTION

31 For the purpose of suspending solid particles in a liquid, mechanically stirred tanks are commonly
32 used in many branches of industry like chemical engineering (Sardeshpande and Ranade, 2012),
33 biotechnology (Trad et al., 2015), and minerals processing (Wu et al., 2011). Typical applications
34 are heterogeneously catalyzed reactions, production of bio-hydrogen, and separation by flotation.
35 In these applications solid particles are suspended in the turbulent flow induced by an impeller,
36 thereby enhancing the solid-liquid heat and mass transfer. The quality of suspension is the result
37 of an intricate interplay between the two phases and the research on this topic has a long and rich
38 history. Next to theoretical and experimental approaches, computational fluid dynamics (CFD)
39 simulation is recently becoming a more and more important means of investigating the
40 hydrodynamics of the solid-liquid flows in all of the mentioned fields of application (Joshi and
41 Nandakumar 2015, Werner et al. 2014, Wang et al. 2018).

42 CFD simulations of solid-liquid flow on the scale of technical equipment are feasible within the
43 Euler-Euler framework of interpenetrating continua combined with the Reynolds-averaged Navier-
44 Stokes (RANS) turbulence models. Since phenomena occurring on the scales of individual particles
45 or groups thereof as well as turbulence are not resolved in this approach, accurate numerical
46 predictions rely on suitable closure relations describing the physics on the un-resolved scales. A
47 large number of works exist, in each of which largely a different and often incomplete set of closure
48 relations is compared to a different set of experimental data. For the limited range of conditions to
49 which each model variant is applied, reasonable agreement with the data is mostly obtained, but
50 due to a lack of comparability between the individual works no complete, reliable, and robust
51 formulation has emerged so far. Moreover, usually a number of empirical parameters are involved
52 and have been adjusted to match the particular data, which deteriorates the applicability.

53 To make a first step towards such a predictive model, we consider adiabatic particulate flows where
54 only momentum is exchanged between the liquid and solid phases, the general approach being
55 similar to a previous investigation on bubbly flows (Shi and Rzehak, 2018). The focus of the work
56 is put on the closures for all interfacial forces acting on particles, which differ significantly from
57 those on bubbles, owing to the different interfacial conditions and deformability (Clift et al., 2005).
58 Cases with low solid fractions, aka dilute suspensions, are considered, where other effects are
59 negligible or at most of secondary importance. Apart from interest in its own right, results obtained
60 for this restricted problem also provide a good starting point for the investigation of more complex
61 situations including flows with moderate to high solids loading (Derksen, 2018), heat and mass
62 transport or gas-solid-liquid three-phase flows (Kim and Kang, 1997). Meanwhile, results obtained
63 should be applicable irrespective of large scale geometry and boundary conditions, as the same
64 closures should work for all systems with same physics at particle scale.

65 The interfacial forces considered here include drag, lift, turbulent dispersion, and virtual mass. The
66 importance of these forces may be summarized as follows. The drag force acts in opposition to the
67 relative motion of a particle with respect to the surrounding fluid and is a key factor in determining
68 the relative velocity of the particles. Virtual mass and turbulent dispersion account for, respectively,
69 the inertia due particle accelerating or decelerating and the interphase turbulent momentum
70 transfer, both of which are likely to be pronounced due to the unsteadiness inherent in stirred-tank
71 flows. The lift force acts perpendicular to both the relative motion and the fluid vorticity. For
72 particles translating within and in parallel to a unidirectional flow the role of lift force is to produce
73 a lateral migration of the particles (Leal, 1980). In Poiseuille flows (either axisymmetric or plane),
74 depending on the flow conditions, the resulting radial profile of solid fraction can peak either near

75 the wall or near the center line. In stirred tank flow, which are highly inhomogeneous, it is difficult
76 to estimate the role of lift force a priori. The ratio between lift and drag may be evaluated from
77 particle tracking simulations (Derksen, 2003, 2012) as $0.2\sqrt{Re_\omega}$ (with Re_ω denoting the shear
78 Reynolds number) indicating a non-negligible lift unless Re_ω is vanishingly small.

79 The paper is organized as follows. In the next section a literature overview of numerical and
80 experimental studies on particulate flows in stirred tanks is given. Section 3 presents all models
81 that are used in this work. Section 4 discusses the selection of test cases from the survey of
82 experiments in section 2 and the numerical issues concerning the present simulations. Section 5
83 presents the main results, i.e. an assessment of several model variants in comparison with the
84 selected test cases. Conclusions and remarks are given in section 6.

85 2 LITERATURE REVIEW

86 2.1 Review of simulation studies

87 Table 1 gives an overview of simulation studies on solid-liquid flow in stirred tanks. Selection of
88 works is based on the following criteria (Shi and Rzehak, 2018): Only works adopting the full
89 Euler-Euler (E-E) or Euler-Lagrange (E-L) frameworks to couple the two-phase flows are taken.
90 Also only works that validate their results by comparison with local measurements are considered.
91 Lastly, for works from each group only the most recent one is listed. As may be seen, in addition
92 to the basic multiphase framework, different modeling options have been chosen concerning
93 turbulence modeling, interfacial forces taken into account, modeling of turbulent dispersion,
94 description of particle-particle interaction, and lastly treatment of impeller rotation.

95 The two frameworks, namely E-E and E-L, differ in the way in which the solid flow is described.
96 In the E-E framework, the solid phase is treated as a continuum with properties analogous to those
97 of a fluid and governed by continuum forms of mass and momentum balance equations. In the E-L
98 framework, particles are tracked individually or as clusters (when the solid fraction is high) based
99 on Newton's second law. The advantage of E-L compared with E-E is that phenomena on the
100 particle scale, such as collisions and particle-fluid interactions, can be represented with greater
101 accuracy. On the other hand, the E-E framework is computationally more efficient for very large
102 systems.

103 Concerning turbulence, the most fundamental approach is direct numerical simulation (DNS).
104 However, this approach is still unfeasible for turbulent flows at industrial scale (Derksen 2012,
105 2018). Two more common approaches are Reynolds averaged Navier-Stokes (RANS) models and
106 large eddy simulation (LES). RANS models can be further divided into two approaches: Reynolds
107 stress models (RSMs) and two-equation eddy-viscosity models. According to Table 1, the latter
108 have been used almost exclusively to study solid-liquid flow in stirred tanks. Due to its assumption
109 of isotropic turbulence this approach generally shows good agreement with the measured data for
110 the mean velocity in the bulk region but fails to predict the flow in regions with strong anisotropy
111 (e.g. the near-impeller region). This limitation may be overcome by anisotropic models such as
112 RSMs. Comparisons between RANS and LES have been performed for both single- and (solid-
113 liquid) two-phase flows (Murthy and Joshi 2008, Guha et al., 2008). From these comparisons the
114 conclusion emerges that, while LES provides improved predictions for single phase flow compared
115 with RANS models, the improvement achieved by LES in two-phase flow predictions is still
116 limited by the models used to couple the phases.

117 For the E-E framework, the turbulence model for the dispersed phase can be dealt with in three
118 basic ways. These are the dispersed model, the mixture model, and the phasic model. The first two
119 use only a single set of equations for, respectively, the continuous phase or the mixture, while the
120 last uses two sets of equations, one for each of the phases. For details we refer to Yang and Mao
121 (2014, section 3.4.3). It should be noted that implementation of the mixture model requires the wall
122 boundary conditions for both phases to be identical. This is physically unreasonable since a viscous
123 fluid cannot slip on the wall while for particles this is typically the case. Notwithstanding,
124 comparisons of alternative approaches (Montante and Magelli, 2005; Fletcher and Brown, 2009;
125 Wadnerkar et al., 2016) have concluded that, for solid fractions below 10%, all three models lead
126 to similar results. Since the phasic model is computationally more expensive, the mixture and the
127 dispersed models are more commonly used as indicated in Table 1.

128 For particulate flows, in addition to the shear-induced turbulence, it is sometimes necessary to take
129 an additional particle-induced turbulence (PIT) into account. According to Table 1, the PIT has
130 mostly been neglected in previous simulations of particulate flow in stirred tanks. In the few works
131 taking it into account two approaches were used. The simpler one is to just add an extra particle-
132 induced contribution to the effective viscosity following Sato et al (1981). To model the PIT effects
133 on TKE and dissipation, additional source terms are introduced directly in the turbulence model
134 equations (Kataoka and Serizawa, 1989).

135 Choice of the particle forces is yet another modeling decision to be made. Among these forces,
136 drag has been assumed to be dominant whereas non-drag forces are often neglected in the works
137 quoted in Table 1. Many correlations for drag force estimation are found in the literature, which
138 are based on particles translating in stagnant liquid (see Loth (2008) for a review). In the highly
139 turbulent flows occurring in stirred tanks, however, it has proven necessary to include the effect of
140 turbulence on the drag force. Models that have been used frequently to account this effect are
141 reviewed in Shah et al. (2015). For works that do take the non-drag forces into account, virtual
142 mass and lift have been frequently considered, while the wall force has been mostly neglected. The
143 virtual mass coefficient is always taken as 0.5, a value that has so far been confirmed to be reliable
144 for dilute systems both numerically and experimentally (see Michaelides and Roig (2011) and
145 references therein). As for the lift, often a positive constant coefficient $C_L = 0.5$ was used
146 (Ljungqvist and Rasmuson, 2001; Ochieng and Lewis, 2006; Guha et al., 2008; Fletcher and Brown,
147 2009). This value is valid for spheres with a free-slip surface in high-Reynolds-number flow but
148 likely not be applicable for solid particles with no-slip surfaces, since the associated lift-generation
149 mechanisms are fundamentally different (Legendre and Magnaudet, 1998).

150 For particulate flows, an important mechanism governing the distribution of particles is the
151 turbulent dispersion, i.e. the transport of the particles by the turbulent eddies. For the E-L
152 framework, this requires the estimation of the instantaneous fluid velocity along the particle
153 trajectory, which is typically modeled by various stochastic approaches (Derksen, 2003;
154 Sommerfeld et al., 2008, section 4.3.3). For the E-E framework, two different approaches are used.
155 In the first, the solid phase continuity equation is augmented by a diffusive term to a convection-
156 diffusion equation (CDE) (see e.g. Loth (2001)). As seen from Table 1, this method has been used
157 quite often. However, there are a significant number of theoretical works which have shown that
158 this simplification might be questionable (Simonin, 1990; Reeks, 1991; Crowe et al., 1996; Drew,
159 2001; Sommerfeld et al., 2008, section 4.4.1). Most of those authors conclude that the essence of
160 dispersion should appear as a force in the momentum equation. So far various formulations
161 regarding this turbulent dispersion force have been proposed, among which the Farve-averaged-

162 drag model (FAD, Burns et al., 2004) and the kinetic theory based model (Reeks, 1991; de
 163 Bertodano, 1998) found numerous applications to particulate flow in stirred tanks (Ljungqvist and
 164 Rasmuson, 2001; Ochieng and Lewis, 2006; Fletcher and Brown, 2009; Qi et al., 2013; Maluta et
 165 al. (2019)).

166 If the system is not dilute particle dispersion is not only caused by turbulence, but in addition also
 167 due to particle-particle collisions. In the E-L framework, this has been dealt with by various
 168 collision models (Derksen, 2003, 2012, 2018). In the E-E framework, the effect of collisions is
 169 included by the kinetic theory of granular flows (KTGF, Gidaspow, 1994), which treats momentum
 170 and energy transfer due particle-particle collisions in the particulate flow in an analogous way as
 171 that for molecules in a single-phase gas. Compared with the various collision models used in the
 172 E-L framework, the significant advantage of the KTGF approach is that there is no need to consider
 173 the mechanical interaction of individual particles so larger systems can be modeled. However, the
 174 constitutive equations needed for the KTGF approach are largely based on empiricism. The
 175 problem of deriving these constitutive equations from more basic physical principles has not yet
 176 been solved and remains a significant challenge for the future.

177 Finally, modeling the flow inside baffled stirred tanks requires suitable boundary conditions for
 178 the impeller blades and the disc on which they are mounted, because these sections are moving
 179 relative to the fixed baffles and the tank wall. Different impeller-rotation models have been
 180 thoroughly described by Yang and Mao (2014, section 3.2.5). We here just note that the most
 181 frequently used approaches include the impeller boundary condition (IBC), sliding mesh/grid (SG),
 182 inner-outer approach (IO), and multiple reference frame (MRF). Comparisons of alternative
 183 modeling approaches have been conducted by Brucato et al. (1998a) and more recently by Shi and
 184 Rzehak (2018). From these comparisons, the MRF has emerged as reliable and considerably more
 185 efficient computationally.

186 Table 1: Simulations of solid-liquid flow in stirred tanks.

Reference	Data source	Multi-phase approach	Turbulence / PIT	Interface forces	Turbulent dispersion	Particle-particle collision	Impeller rotation *)
Kohnen (2000)	Kohnen (2000)	E-E	dispersed $k-\varepsilon$ / none	F^{drag}	none	KTGF	SG
Ljungqvist & Rasmuson (2001)	Ljungqvist & Rasmuson (2004)	E-E	phasic $k-\varepsilon$ / none	$F^{\text{drag}}, F^{\text{lift}}, F^{\text{VM}}$	none / force	none	IBC
Oshinowo & Bakker (2002)	Godfrey & Zhu (1994), Guiraud et al. (1997)	E-E	dispersed $k-\varepsilon$ / none	F^{drag}	none	KTGF	IBC
Wang et al. (2003)	Nouri & Whitelaw (1992), Yamazaki et al. (1986)	E-E	dispersed $k-\varepsilon$ / source terms	F^{drag}	CDE	none	IO
Khopkar et al. (2006)	Yamazaki et al. (1986), Godfrey & Zhu (1994)	E-E	mixture $k-\varepsilon$ / none	F^{drag}	CDE	none	MRF

Montante & Magelli (2007)	Montante & Magelli (2007)	E-E	mixture $k-\varepsilon$ / none	F^{drag}	CDE	none	SG
Guha et al. (2008)	Guha et al. (2007)	E-E	mixture $k-\varepsilon$ / none	$F^{\text{drag}}, F^{\text{lift}}, F^{\text{VM}}$	force	KTGF	MRF
		E-L	LES / none		stochastic tracking	included	IBC
Kasat et al. (2008)	Yamazaki et al. (1986)	E-E	mixture $k-\varepsilon$ / none	F^{drag}	force	none	MRF
Ochieng & Onyango (2008)	Ochieng & Lewis (2006)	E-E	dispersed $k-\varepsilon$ / Sato	$F^{\text{drag}}, F^{\text{lift}}, F^{\text{VM}}, F^{\text{wall}}$	force	KTGF	SG
Shan et al. (2008)	Shan et al. (2008)	E-E	dispersed $k-\varepsilon$ / source terms	F^{drag}	none	none	IBC
Sardeshpande et al. (2011)	Sardeshpande et al. (2011)	E-E	mixture $k-\varepsilon$ / none	F^{drag}	force	none	MRF
Feng et al. (2012)	Yamazaki et al. (1986), Micheletti et al. (2003, 2004), Guha et al. (2007), Montante et al. (2012)	E-E	dispersed $k-\varepsilon$ & RSM / source terms	F^{drag}	force	none	IO
Liu & Barigou (2014)	Liu & Barigou (2014)	E-E	mixture $k-\varepsilon$ / none	F^{drag}	none	none	MRF
Tamburini et al. (2014)	Micheletti et al. (2003)	E-E	dispersed & mixture $k-\varepsilon$ / none	F^{drag}	force / CDE	none	MRF / SG
Wadnerkar et al. (2016)	Guida et al. (2010)	E-E	dispersed, mixture, and phasic $k-\varepsilon$ & RSM / none	F^{drag}	force	none / KTGF	MRF
Wang et al. (2017)	Pianko-Oprych et al. (2009)	E-E	dispersed $k-\varepsilon$ / none	$F^{\text{drag}}, F^{\text{VM}}$	none	KTGF	MRF
Li et al. (2018)	Li et al. (2018)	E-L	DNS	F^{drag}	resolved	included	IBC
Maluta et al. (2019)	Carletti et al. (2014)	E-E	mixture $k-\varepsilon$ & RSM	$F^{\text{drag}}, F^{\text{lift}}$	force	none / KTGF	MRF

187 *) IO, inner–outer method; SG, sliding grid/mesh; IBC, impeller boundary condition; MRF, multiple reference frame.
188 Other items: CDE, convection diffusion equation; RSM, Reynolds stress model; KTGF, kinetic theory of granular
189 flows. Further explanations are given in the text.

190 2.2 Review of experimental studies

191 An overview of previously reported experimental studies on solid-liquid flow in mechanically
192 stirred tanks is shown in Table 2. The focus is on works that provide measurements of spatially
193 resolved data for monodisperse suspensions. Finally, for measurements conducted by the same
194 group and employing identical techniques, only the most recent work is listed. Exceptions to this
195 last rule are works that have been used for comparison in the simulation studies above.

196 For most of the experimental studies, a single standard Rushton turbine or pitched blade turbine
197 rotating with roughly 200 to 1200 rpm is used, the tank diameter is in the range of 100 to 500 mm
198 and the ratio of fill height to diameter is close to one. For works using multiple impellers (Magelli
199 et al., 1990; Montante et al., 2002; Montante and Magelli, 2007), the aspect ratio is increased in
200 proportion. Bigger tanks are considered by Spidla et al. (2005) and Angst and Kraume (2006),
201 smaller ones by Gabriele et al. (2011).

202 Most works listed in Table 2 focus on the so-called complete suspension condition (i.e. conditions
203 with an impeller rotation speed much higher than the just-suspension speed (see Guha et al. (2007)
204 and references therein)). Cases with incomplete suspension are also investigated in Nouri and
205 Whitelaw (1992), Micheletti et al. (2003), Tamburini et al. (2013), and Carletti et al. (2014). The
206 glass-water system with a solid-to-liquid density ratio of ≈ 2.5 has been investigated quite often.
207 For lower density ratios, polystyrene or polymethylmethacrylate (PMMA, e.g. Diakon™) particles
208 were used (Magelli et al., 1990; Nouri and Whitelaw, 1992; Micheletti et al., 2003, 2004; Montante
209 and Magelli, 2007; Gabriele et al., 2011; Sardeshpande et al., 2011) while higher density ratios are
210 obtained for bronze (Magelli et al., 1990; Montante and Magelli, 2007) or nickel particles
211 (Ljungqvist and Rasmuson, 2004; Ochieng and Lewis, 2006). Various aqueous solutions with
212 identical refractive index as that of the suspended solid phase are sometimes selected as the working
213 fluid, while their densities are always comparable to that of water. The investigated mean
214 (volumetric) solid fraction spans a wide range of 0.1% to 30%. Significantly lower solid loadings
215 are considered by Ljungqvist and Rasmuson (2004, 0.01%) and Tamburini et al. (2013, $< 0.01\%$),
216 respectively. The particle diameter is mostly in the range of 0.1 to 1 mm. Coarser particles are
217 considered by Gabriele et al. (2011, 1.5 mm), Pianko-Oprych et al. (2009, 3 mm), Guida et al.
218 (2010, 3 mm), and Li et al. (2018, 8 mm).

219 As for the data, an ideal data set that contains all relevant observables (i.e. phase mean and
220 fluctuation velocities as well as solid fraction) with high spatial resolution and profiles along
221 several directions at several positions is available so far only from the data sets of Nouri and
222 Whitelaw (1992) and Unadkat et al. (2009). However, the image analysis method used by Unadkat
223 et al. (2009) might not be reliable (Tamburini et al., 2013) and the resulting fraction maps obtained
224 should be interpreted as an indication only. Some relatively comprehensive data sets, e.g. Guiraud
225 et al. (1997), Ochieng and Lewis (2006), and Chen et al. (2011), employed complex-shaped
226 impellers whose geometry is unfortunately not fully specified. Besides, although the experiment of
227 Nouri and Whitelaw (1992) considered varying values of density ratio, mean solid fraction, and
228 particle diameter only one data set provides the information of both mean velocity and local solid
229 fraction (see Table 6 for the details). Thus to achieve a solid validation, a combination of several
230 data sets seems necessary.

231 Measurement methods are partly intrusive using various well-known probe techniques, like
232 electrical and optical needle probes (denoted as IP and OP in Table 2), for solid fraction.
233 Photographic methods to determine solid fraction by image analysis (IA) have been adapted for

234 use in stirred tanks using either backlighting (Magelli et al., 1990) or laser light sheets (Unadkat et
 235 al., 2009; Tamburini et al., 2013). Methods like PIV, LDA, and PDA can readily be used to measure
 236 particle velocity and by adding tracer particles also liquid velocity. These optical techniques are
 237 non-intrusive but limited to suspensions with low solids loading. This drawback can be overcome
 238 by matching the refractive index of the liquid to that of the dispersed phase (RIM) as several works
 239 in Table 2 have shown. More sophisticated techniques that allow non-intrusive probing of opaque
 240 suspensions are radioactive particle tracking techniques including CARPT and PEPT. Both resolve
 241 directly Lagrangian particle trajectories while the Eulerian information like phase velocity and
 242 fraction is obtained by applying appropriate reconstructing algorithms. More advanced
 243 tomographic methods such as electrical resistance tomography (ERT) and ultrasound velocity
 244 profiling (UVP) are just about beginning to be applied to this field.

245

Table 2: Experiments on solid-liquid flow in stirred tanks.

Reference	Tank diameter / Fill height (mm)	Impeller type *) / Diameter / Bottom clearance (mm)	Density ratio / Solid fraction (v/v %) / Particle diameter (mm)	Rotation rate (rpm)	Technique **)	Measured quantities
Yamazaki et al. (1986)	300 / 300	RT / 70 / 90	2.37 - 2.62 / 5, 20 / 0.087 - 0.23	300 - 1200	OP	$\bar{\alpha}_S$
Magelli et al. (1990)	43.5 / 174 78.7 / 315	4×RT / 14.5 / 21.8 4×RT / 26.2 / 39.4	1.02-8.41 / ~ 7.5 / 0.14 - 0.98	302 - 1008	IA	$\bar{\alpha}_S$
Nouri & Whitelaw (1992)	294 / 294	RT / 98, 147 / 73.5, 98	1.18 - 2.95 / 0, 0.02 - 2.5 / 0.23 - 0.67	150 - 313	LDA & RIM	$\bar{u}_L, u'_L, \bar{u}_S, u'_S, \bar{\alpha}_S$
Godfrey & Zhu (1994)	154 / 154	PBT / 51 / 46	2.26 / 0.4 - 30 / 0.23 - 0.67	600 - 1600	RIM	$\bar{\alpha}_S$
Guiraud et al. (1997)	300 / 300	M-TT / 140 / 100	2.23 / 0, 0.5 / 0.25	306	PDA	$\bar{u}_L, u'_L, \bar{u}_S, u'_S$
Kohnen (2000)	220 / 220	RT / 94 / 73.3	~ 2.5 / 0, 5, 10 / 0.55	650	LDA & RIM	\bar{u}_L, u'_L
Montante et al. (2002)	230 / 920 480 / 1440	4×PBT / 94 / 115 3×PBT / 195 / 240	2.45 / 0.12 - 0.41 / 0.13 - 0.79	486 - 1200	OP	$\bar{\alpha}_S$
Micheletti et al. (2003)	290 / 290	RT / 98 / 43.5-96.6	1.05 - 2.47 / 0, 1.8 - 15.5 / 0.15 - 0.71	100 - 1200	IP	$\bar{\alpha}_S$
Ljungqvist & Rasmuson (2004)	297 / 297	PBT / 99 / 99	2.45 - 8.9 / 0, ~ 0.01 / 0.14 - 0.45	180 - 540	PDA	\bar{u}_L, \bar{u}_S
Micheletti & Yianneskis (2004)	80.5 / 80.5	RT / 27 / 27	1 / 0, 0.1 - 2.0 / 0.19	2500	LDA & RIM	\bar{u}_L, u'_L
Spidla et al. (2005)	1000 / 1000	PBT / 333 / 167, 333	2.5 / 5, 10 / 0.14, 0.35	156 - 267	IP	$\bar{\alpha}_S$
Angst & Kraume (2006)	200 / 200 400 / 400 900 / 900	PBT 62.5 / 62.5 PBT 125 / 125 PBT 281 / 281	2.5 / 2 - 10 / 0.2	678, 877 419, 538 275	OP	$\bar{\alpha}_S$
Ochieng & Lewis (2006)	380 / 380	M-HA / 126.7 / 57	8.9 / 0, 0.03 - 2 / 0.15 - 1.0	200 - 500	LDA & IA	$\bar{u}_L, \bar{\alpha}_S$
Montante & Magelli (2007)	232 / 928	4×RT / 79 / 116	1.15, 2.46 / 0.05 - 0.15 / 0.33	1146, 1457	PIV & IA	$\bar{u}_L, \bar{\alpha}_S$
Guha et al. (2007)	200 / 200	RT / 66.7 / 66.7	2.5 / 1, 7 / 0.3	850 - 1200	CARPT	\bar{u}_S, u'_S
Virdung & Rasmuson (2007)	150 / 150	PBT / 50 / 50	2.5 / 0, 0.5 - 1.5 / 1.0	900	PIV & RIM	\bar{u}_L, \bar{u}_S
Shan et al. (2008)	300 / 420	PBT / 80 / 160	1.97 / 0.5 / 0.08	113 - 173	OP	$\bar{\alpha}_S$

Pianko-Oprych et al. (2009)	288 / 288	PBT / 144 / 72	2.16 / 0, 2.31 / 3.0	150 - 406	PEPT	\bar{u}_L, \bar{u}_S
Unadkat et al. (2009)	101 / 101	PBT / 33.7 / 25.25	2.5 / 0, 0.2 - 0.5 / 1.0	1600	PIV & IA	$\bar{u}_L, \bar{u}_S, u'_L, u'_S, \varepsilon_L, \bar{\alpha}_S$
Guida et al. (2010)	288 / 288	PBT / 144 / 72	2.16 / 0, 2.5 - 23.6 / 3.0	330 - 590	PEPT	$\bar{u}_L, \bar{u}_S, \bar{\alpha}_S$
Chen et al. (2011)	220 / 220	CBY / 139 / 55	2.50 / 0, 0.2 - 5.0 / 0.65	410	PIV& RIM	$\bar{u}_L, u'_L, \varepsilon_L$
Gabriele et al. (2011)	45 / 45	PBT / 24.5 / 15	1.38 / 0, 1.5, 5.0 / 1.5	900	PIV& RIM	$\bar{u}_L, u'_L, \varepsilon_L$
Sardeshpande et al. (2011)	700 / 700	PBT / 200 / 233	1.06, 2.5 / 1 - 7 / 0.25, 0.35	202 - 275	UVP	\bar{u}_L, \bar{u}_S
Harrison et al. (2012)	220 / 220	RT / 110 / 77.3	2.65 / 5, 10, 20 / 0.16, 0.51, 0.73	236, 547	ERT	$\bar{\alpha}_S$
Montante et al. (2012)	232 / 232	RT / 77.3 / 77.3	2.47 / 0, 0.05 - 0.20 / 0.12 - 0.77	852	PIV	\bar{u}_L, u'_L
Tamburini et al. (2013)	190 / 190	RT / 95 / 63.3	2.48, 3.45 / 0.006, 0.008 / 0.13, 0.5	300 - 600	IA	$\bar{\alpha}_S$
Carletti et al. (2014)	232 / 250	PBT / 78 / 78	2.5 / 9 - 15 / 0.13, 0.37	500 - 900	ERT	$\bar{\alpha}_S$
Gu et al. (2017)	480 / 800	PBT+RT / 200 / 160	2.47 / 5 / 0.12	60 - 380	Sampling	$\bar{\alpha}_S$
Li et al. (2018)	220 / 220	PBT / 158 / 44	1.63, 2.21 / 0, 1 - 8 / 8	450, 496	PIV& RIM	$\bar{u}_L, u'_L, \bar{\alpha}_S$

246 *) CBY, down-pumping 3-blade propeller; M-HA, Mixtec HA735 propeller; M-TT, Mixel TT propeller; PBT, pitched
247 blade turbine; RT, Rushton turbine.

248 **) Intrusive: IP, impedance probe; OP, optical probe. Non-intrusive: CARPT, computer automated radioactive
249 particle tracking; ERT, electrical resistance tomography; IA, image analysis; LDA, laser Doppler anemometry; PDA,
250 phase-Doppler anemometry; PEPT, positron emission particle tracking; PIV, particle image velocimetry; RIM,
251 refractive index matching; UVP, ultrasound velocity profiler.

252 2.3 Reynolds numbers and lengthscales

253 The review above facilitates to evaluate roughly the ranges of parameters that apply to solid
254 suspensions in stirred tank flows covered in experiments. These include in particular various
255 lengthscales, i.e. the particle size, the Kolmogorov lengthscale, and the typical size of the energy
256 containing eddies, as well as the relative velocity between the particles and the liquid. From these,
257 particle- and shear Reynolds numbers may be derived. Ranges of these parameters are important
258 for the development of closure models of, especially, the interfacial forces.

259 The particle Reynolds number is defined as $Re_p = d_p u_{rel} / \nu$, where d_p is the particle diameter,
260 u_{rel} denotes the magnitude of relative velocity, and ν is the liquid kinematic viscosity. According
261 to the experiments listed in Table 2, the glass-water system has been frequently considered. In this
262 case the kinematic viscosity of the liquid ν is about $10^{-6} \text{ m}^2\text{s}^{-1}$. The ratio of the particle density
263 ρ_S to that of the liquid phase ρ_L is around 2.5. The typical size of the particles considered is in the
264 range $0.1 \text{ mm} \leq d_p \leq 1 \text{ mm}$. The terminal settling velocity is most often used as a reference for
265 the relative velocity. In the Stokes limit, the settling velocity is $d_p^2 g (\rho_S - \rho_L) / (18 \nu \rho_S)$ indicating
266 $1 \leq Re_p \leq 1000$. This estimation of Re_p can be improved by considering two aspects. On one
267 hand, the finite Reynolds number effect is to increase the Stokes drag by a ratio of
268 $(1 + 0.15 Re_p^{0.687})$ (Schiller and Naumann, 1933) and thus to reduce the relative velocity. On the
269 other hand, flow in a stirred tank is highly turbulent such that particle inertia significantly contribute

270 to the relative velocity. Previous simulation results (Derksen, 2003, 2012; Khopkar et al., 2006;
 271 Guha et al., 2008) indicate that the magnitude of relative velocity in the near impeller region is
 272 approximately 2 times that of the settling velocity. The combination of both aspects results in a
 273 somewhat narrower range of $1 \leq Re_p \leq 800$.

274 The shear Reynolds number is defined as $Re_\omega = d_p^2 \omega / \nu$ with ω denoting the magnitude of flow
 275 shear rate. In stirred tank flows, ω is proportional to the impeller rotation rate Ω (in rev/s). CFD
 276 simulations (Derksen and Van den Akker, 1998; Derksen, 2003) indicate that ω easily exceeds
 277 10Ω in the near impeller region, which can be taken as an upper estimate. According to Table 2, Ω
 278 has a magnitude around 10 which gives a range of $0 < Re_\omega \leq 100$.

279 The Kolmogorov lengthscale η for stirred tank flows with fully developed turbulence can be
 280 estimated empirically (Derksen, 2003, 2012) as $\eta = D_i Re_i^{-0.75}$, where D_i is the impeller diameter
 281 and $Re_i = \Omega d_i^2 / \nu$ denotes the impeller Reynolds number. According to Table 2, the impeller
 282 Reynolds number is around 5×10^4 , and the impeller diameter is around 0.1 m. Thus η has a
 283 magnitude of $O(10^{-2})$ mm which agrees with values estimated in DNS studies (Gillissen and Van
 284 den Akker, 2012; Derksen, 2012). On the other hand, the Eulerian longitudinal integral lengthscale
 285 Λ , which is a measure of the energy-containing eddies, should be about the same order of magnitude
 286 as turbulence-generating sources. For stirred tank flows, the impeller blade was suggested to be the
 287 major turbulence source (Wu and Patterson, 1989). The typical size of the impeller blade is
 288 $1/15 \sim 1/12$ that of the tank diameter thus Λ has a magnitude of $O(10^1)$ mm.

289 3 OVERVIEW OF MODELS

290 This section describes the simulation models employed. Section 3.1 briefly summarizes the basic
 291 conservation equations of the E-E framework, which is applied in the present work. Since various
 292 particle forces are known from previous works to have an effect on the accuracy of the model
 293 predictions, an attempt is made here to assemble a rather complete description of these forces,
 294 which is detailed in section 3.2. Section 3.3 discusses effects of liquid phase turbulence on the
 295 particles which comprise a modification of the drag force due to turbulence and the modeling of
 296 turbulent dispersion. The modeling of turbulence in the liquid phase is based on the Reynolds stress
 297 model proposed by Speziale, Sarkar, and Gatski (Speziale et al., 1991, hereafter referred to as SSG
 298 RSM), which has been used successfully in previous work and is described in section 3.4.
 299 Comparisons of different RANS models for stirred tank simulations are provided for example in
 300 Ciofalo et al. (1996), Cokljat et al. (2006), Murthy and Joshi (2008), Feng et al. (2012), Morsbach
 301 (2016), Wadnerkar et al. (2016), and Shi and Rzehak (2018).

302 3.1 Euler-Euler framework for solid-liquid flow

303 Using the index $k = L, S$ to denote the liquid and solid phase, respectively, the phasic continuity
 304 and Navier–Stokes equations take the form (Drew and Passman, 2006)

$$\frac{\partial}{\partial t}(\alpha_k \rho_k) + \nabla \cdot (\alpha_k \rho_k \mathbf{u}_k) = 0 \quad (1)$$

305 and

$$\frac{\partial}{\partial t}(\alpha_k \rho_k \mathbf{u}_k) + \nabla \cdot (\alpha_k \rho_k \mathbf{u}_k \otimes \mathbf{u}_k) = \quad (2)$$

$$-\alpha_k \nabla p_k + \nabla \cdot (2\alpha_k \mu_k^{\text{mol}} \mathbf{D}_k) - \nabla \cdot (\alpha_k \rho_k \mathbf{R}_k) + \mathbf{F}_k^{\text{body}} + \mathbf{F}_k^{\text{inter}}.$$

306 In Eq. (2), α is the volume fraction, p denotes the pressure, $\mathbf{D} = (\nabla \mathbf{u} + (\nabla \mathbf{u})^T)/2$ is the strain rate
 307 tensor, and μ^{mol} is the molecular dynamic viscosity. μ_S^{mol} is assumed to be identical with μ_L^{mol} , an
 308 assumption that was made in most simulation studies listed in Table 1. \mathbf{R} is the Reynolds stress
 309 tensor which is defined in terms of the turbulent fluctuating velocities \mathbf{u}'_k as $\mathbf{R}_k = \langle \mathbf{u}'_k \otimes \mathbf{u}'_k \rangle$,
 310 where $\langle \ \rangle$ makes the involved averaging operation explicit. \mathbf{R}_L is obtained by directly solving a
 311 transport equation as discussed in detail in section 3.4 while \mathbf{R}_S is presently neglected.

312 The body forces $\mathbf{F}_k^{\text{body}}$ comprises the gravity force as well as centrifugal and Coriolis forces where
 313 a rotating frame of reference is adopted.

314 The term $\mathbf{F}_k^{\text{inter}}$ accounts for the momentum transfer between the phases. Due to momentum
 315 conservation the relation $\mathbf{F}_S^{\text{inter}} = -\mathbf{F}_L^{\text{inter}}$ holds. This term comprises of a number of contributions
 316 and the corresponding models employed here are summarized in Table 3. A detailed discussion
 317 thereof will be given in sections 3.2 and 3.3.

318 Table 3: Summary of particle force correlations.

force	reference
drag	Schiller and Naumann (1933) with modification due to turbulence discussed in section 3.3.2
lift	Shi and Rzehak (2019)
turbulent dispersion	de Bertodano (1998) with turbulence scales discussed in section 3.3.1
virtual mass	constant coefficient $C_{\text{VM}} = 1/2$

319

320 3.2 Interfacial forces

321 Interfacial forces considered include drag, lift, virtual mass, and turbulent dispersion. Although the
 322 last one has been frequently classified as an interfacial force, its description is deferred to section
 323 3.3.3, because it depends on turbulence parameters that are naturally introduced only in section
 324 3.3.1. For flow in the near wall region there could be additional wall effects, e.g. an enhancement
 325 of drag (Sommerfeld et al., 2008, section 3.1) or a suppression of shear-lift (Shi and Rzehak, 2020).
 326 Moreover, the presence of the wall introduces a wall-lift force directed away from the wall (Shi
 327 and Rzehak, 2020). These wall effects are important for modeling multiphase flows in rather
 328 confined geometries but are neglected in the present study as the near-wall region occupies only a
 329 small portion of the stirred tank volume.

330 3.2.1 Drag force

331 The drag force acting on the dispersed phase takes the form

$$\mathbf{F}_S^{\text{drag}} = -C_D \frac{3}{4} d_p^{-1} \rho_L \alpha_S u_{\text{rel}} \mathbf{u}_{\text{rel}}, \quad (3)$$

332 where $\mathbf{u}_{\text{rel}} = \mathbf{u}_S - \mathbf{u}_L$ denotes the relative velocity and C_D is the drag coefficient. For solid spheres
 333 translating in a stagnant fluid, the drag correlation of Schiller and Naumann (1933) applies:

$$C_{D,0} = \frac{24}{Re_p} (1 + 0.15 Re_p^{0.687}). \quad (4)$$

334 3.2.2 Lift force

335 The lift force acting on the dispersed phase takes the form

$$\mathbf{F}_S^{\text{lift}} = C_L \alpha_S \rho_L \boldsymbol{\omega}_L \times \mathbf{u}_{\text{rel}}, \quad (5)$$

336 where $\boldsymbol{\omega}_L$ gives the vorticity of the fluid with $\boldsymbol{\omega}_L \equiv \nabla \times \mathbf{u}_L$, and C_L is the lift coefficient. The lift
337 force for spherical particles rotating freely only under the action of the surrounding fluid with no
338 external torque applied is frequently described by a linear combination of contributions from flow
339 vorticity and particle rotation (Shi and Rzehak, 2019), i.e.

$$C_L = C_{L\omega} + \frac{3}{8} f_{\omega-\Omega} C_{L\Omega}, \quad (6)$$

340 where $C_{L\omega}$ and $C_{L\Omega}$ denote the coefficients of the vorticity- and rotation-induced lift forces,
341 respectively. $f_{\omega-\Omega}$ is the dimensionless vorticity-induced rotation rate defined by $f_{\omega-\Omega} \equiv 2\Omega_{fr}/\omega$,
342 where Ω_{fr} denotes the vorticity-induced particle rotation rate in the torque-free condition.

343 According to the review of Shi and Rzehak (2019) the two lift coefficients take the form:

$$C_{L\omega} = \begin{cases} \frac{27}{2\pi^2} (Sr Re_p)^{-1/2} J(\epsilon) - \frac{33}{32} \exp(-0.5 Re_p) & \text{for } Re_p \leq 50 \\ -0.048 Sr^{-1} \exp(0.525 Sr) \left\{ 0.49 + 0.51 \tanh \left[5 \lg \left(\frac{Re_p Sr^{0.08}}{120} \right) \right] \right\} & \text{for } Re_p > 50 \end{cases} \quad (7)$$

344 and

$$C_{L\Omega} = 1 - 0.62 \tanh(0.3 Re_p^{1/2}) - 0.24 \tanh(0.01 Re_p) \coth(0.8 Rr^{0.5}) \arctan[0.47(Rr - 1)], \quad (8)$$

345 where Sr and Rr denote, respectively, the dimensionless flow vorticity with $Sr = \omega d_p / u_{\text{rel}}$ and
346 the dimensionless particle rotation rate with $Rr = \Omega_{fr} d_p / u_{\text{rel}}$, $J(\epsilon)$ is the function defined by
347 McLaughlin (1991, Eq. (20)), and ϵ is a lengthscale ratio defined by $\epsilon = \sqrt{Sr / Re_p}$. An appropriate
348 correlation for $J(\epsilon)$ was proposed by Legendre and Magnaudet (1998) as

$$J(\epsilon) = 2.255(1 + 0.20\epsilon^{-2})^{-3/2}. \quad (9)$$

349 $f_{\omega-\Omega}$ in Eq. (6) is related to the particle- and shear Reynolds numbers via (Shi and Rzehak, 2019)

$$f_{\omega-\Omega} = \{1 + 0.4[\exp(-0.0135 Re_\omega) - 1]\} (1 - 0.07026 Re_p^{0.455}). \quad (10)$$

350 Eqs. (5) - (10) summarize the lift force correlation proposed by Shi and Rzehak (2019) concerning
351 solid particles translating in stream-wise linear shear flows under torque-free conditions. Its
352 advantage over the older correlation of Mei (1992), which has been widely used in engineering,
353 lies in two aspects. Firstly it accounts for the contributions from flow vorticity and particle rotation
354 simultaneously while the correlation in Mei (1992) accounts for the former only. DNS studies have
355 shown the necessity to consider the (torque-free) rotation-induced lift when $Re_p \geq 5$. Secondly,
356 the correlation in Mei (1992) neglects negative values of $C_{L\omega}$, which have been revealed in DNS
357 studies beyond $Re_p = 50$. This effect is taken into account in the correlation from Shi and Rzehak
358 (2019). These advantages motivate application of the correlation of Shi and Rzehak (2019) to
359 describe the lift force.

360 3.2.3 Virtual mass force

361 The virtual mass force acting on the dispersed phase takes the form

$$\mathbf{F}_S^{\text{VM}} = C_{\text{VM}} \alpha_S \rho_L \left(\frac{D_L \mathbf{u}_L}{Dt} - \frac{D_S \mathbf{u}_S}{Dt} \right), \quad (11)$$

362 where D_L/Dt and D_S/Dt denote material derivatives with respect to the liquid and solid velocities,
363 respectively. For the virtual mass coefficient a value of $C_{\text{VM}} = 0.5$ is applied.

364 **3.3 Turbulence effects**

365 This section discusses the turbulence effects on the interfacial forces. Since most of the turbulence-
366 particle interactions are depend on the particle-turbulence interaction timescale, this quantity is
367 discussed first. Modeling of the drag modification and turbulent dispersion force are then
368 discussed.

369 3.3.1 Particle-turbulence interaction timescale

370 The particle-turbulence interaction timescale T_L^S is a crucial parameter in describing turbulence
371 effects on the motion of the dispersed phase (Balachandar and Eaton, 2010). A simple form of this
372 timescale (Loth, 2001) accounting for the crossing-trajectories effect (Yudine, 1959) is composed
373 of the Lagrangian integral timescale following the fluid motion, T_L^L , and the time for a particle to
374 cross an typical eddy, τ_{cross} , as:

$$T_L^S \approx \left(T_L^L{}^{-2} + \tau_{\text{cross}}^{-2} \right)^{-\frac{1}{2}}. \quad (12)$$

375 This simple form neglects the continuity effect (Csanady, 1963), accounting for which however
376 would make it necessary to describe T_L^S in tensor form. To avoid this complication here as well as
377 in the modeling of turbulent dispersion, the scalar form Eq. (12) is employed presently as a basic
378 description. The two timescales are defined in terms of quantities that can be computed from a
379 RANS turbulence model as

$$\begin{aligned} T_L^L &= C_T \frac{k}{\varepsilon} \\ \tau_{\text{cross}} &= \frac{\Lambda}{u_{\text{rel}}} = C_\Lambda \frac{k^{3/2}}{u_{\text{rel}} \varepsilon} \end{aligned} \quad (13a, b)$$

380 where Λ denotes the Eulerian longitudinal integral lengthscale and $u_{\text{rel}} = |u_S - u_L|$ the relative
381 velocity. The constants C_T and C_Λ are discussed in the following.

382 According to the cornerstone dissipation scaling of turbulence, sometimes referred to as
383 Kolmogorov's zeroth law (Pearson et al., 2004),

$$\Lambda = C_\varepsilon \frac{(2/3 k)^{3/2}}{\varepsilon}. \quad (14)$$

384 Here C_ε is a constant which is expected to be universal in the limit of high Reynolds numbers.
385 While the verification of this assertion and the determination of the numerical value of C_ε is still
386 an active subject of research, quite a few studies (reviewed by Ishihara et al. (2009)) have shown
387 that $C_\varepsilon \approx 0.43$ for simulations of statistically stationary forced turbulence in a periodic box. The
388 same value has also been obtained by Pope (2000, sect. 6.5.7) from a model for the turbulent energy

389 spectrum. Using this value in Eq. (14) and comparing with Eq. (13b) gives $C_\Lambda \approx 0.234$. In the
 390 present work, this value will be used since at least it is well-established for a well-defined
 391 idealization. In the absence of clear and unambiguous results for more realistic situations this
 392 provides the best available starting point, on which future improvements may be based. However,
 393 it has to be acknowledged that the value of C_ε depends on initial and boundary conditions
 394 (Vassilicos, 2015), e.g. twice as high values are often found experimentally in grid-generated
 395 turbulence. Values assumed for C_Λ in previous studies of particulate flows, often with little to no
 396 further justification, range between 0.09 and 0.54 (see Table 4).

397 Results concerning T_L^L are mostly presented in terms of the dimensionless parameter

$$\beta = \frac{T_L^L (2/3 k)^{1/2}}{\Lambda} = T_L^L \frac{(2/3)^{1/2} \varepsilon}{C_\Lambda k'} \quad (15)$$

398 by virtue of Eq. (13b). Solving for T_L identifies

$$C_T = \frac{C_\Lambda}{(2/3)^{1/2}} \beta \approx 0.287\beta \quad (16)$$

399 using the value of C_Λ from above. Simulations of statistically stationary forced turbulence in a
 400 periodic box give an asymptotic value of $\beta \approx 0.78$ at large Reynolds numbers (Yeung et al., 2001;
 401 Sawford et al., 2008; Sawford and Yeung, 2011), which corresponds to $C_T \approx 0.224$. Like above,
 402 this well-defined value will be used as a starting point in the present work. Again it has to be
 403 acknowledged that experiments on grid turbulence often show values of β as low as half of the one
 404 used here. Values for C_T assumed in previous studies of particulate flows are given in Table 4,
 405 from which a wide spread of values ranging from 0.09 to 0.5 becomes obvious. Thus, a systematic
 406 study of the influence of different choices seems appropriate.

407 Table 4: Values of C_T and C_Λ used in previous studies.

reference	C_T^\dagger	C_Λ	β
Snyder and Lumley (1971)	-	-	~ 0.92
Tennekes and Lumley (1971, Eq. (8.5.15))	-	-	$2/3 = 0.67$
Shlien and Corrsin (1974, $R_\lambda \approx 70$)	-	-	1
Calabrese and Middleman (1979)	0.41	-	-
Boysan et al. (1982)	$0.5 \times 2^{-1/2} C_\mu^{3/4} = 0.058$	-	-
Gosman and Loannides (1983)	-	$C_\mu^{1/2} = 0.3$	-
Pourahmadi and Humphrey (1983)	0.41	-	-
Shuen et al. (1983)	$0.5 \times (3/2)^{1/2} C_\mu^{3/4} = 0.101$	$C_\mu^{3/4} = 0.164$	-
Chen and Wood (1984)	$C_\mu^{3/4} = 0.164$	$C_\mu^{3/4} = 0.164$	-

[†] Note that the relation between the ‘‘eddy life-time’’, which has been frequently referred to in references listed in Table 4, and the Lagrangian intergral time scale T_L^L depends on the functional shape of the Lagrangian velocity correlation coefficient (see Gouesbet and Berlemont (1999) for details).

Mostafa and Mongia (1987)	$(3/2)^{1/2}C_\mu^{3/4} = 0.201$	$C_\mu^{3/4} = 0.164$	1
Sato and Yamamoto (1987, $R_\lambda = 70$)	-	-	0.3 - 0.6
Amsden et al. (1989, page 17)	0.50	$C_\mu^{3/4} = 0.164$	-
Simonin and Viollet (1990)	$(3/2)^{1/2}C_\mu = 0.110$	$(3/2)^{1/2}C_\mu = 0.110$	-
Zhou and Leschziner (1991)	$0.8 \times (3/2)^{1/2}C_\mu^{3/4} = 0.161$	-	-
Lu (1995)	-	$(0.212/0.36) \times (3/2)^{1/2} = 0.32$	0.36
de Bertodano (1998)	$C_\mu^{3/4} = 0.164$	$1/2 C_\mu^{1/4} = 0.274$	-
Peirano and Leckner (1998)	$C_\mu = 0.09$	$C_\mu = 0.09$	-
Sreenivasan (1998)	-	$(2/3)^{3/2} \times 0.424 = 0.231$	-
Loth (2001)	0.27	$1.6C_\mu^{3/4} = 0.263$	-
Yeung et al. (2001, 2006, $38 \leq R_\lambda \leq 648$)	-	-	0.78
Sommerfeld et al. (2008, section 4.3.3)	0.24	$(2/3)^{1/2} \times 0.3 = 0.245$	-
Ishihara et al. (2009)	-	$(2/3)^{3/2} \times 0.43 = 0.234$	-
Sawford and Yeung (2011, $38 \leq R_\lambda \leq 1000$)	-	-	0.74
Vassilicos (2015)	-	0.218 ... 0.544	-

408

409 3.3.2 Drag modification

410 The high turbulence intensity of flow in stirred-tanks has an appreciable effect on the mean drag
411 force acting on the suspended particles. The recent review of Balachandar and Eaton (2010) shows
412 that different mechanisms may be active and different phenomena are observed depending on the
413 precise conditions, but a comprehensive understanding has not been achieved yet. For mechanically
414 agitated dilute suspensions of particles, experimental work summarized by Fajner et al. (2008)
415 shows that the settling velocity is typically smaller than that in a still fluid, which indicates an
416 increase in the apparent drag force due to turbulence. The problem involves several relevant
417 parameters (Good et al., 2014), most prominently the Stokes number St , i.e. the ratio of particle
418 and turbulence timescales.

419 A quantitative model for the modification factor of the drag force due to turbulence was developed
420 by Lane et al. (2005). Denoting the terminal velocity u_{term} in still fluid by an index “0” and that
421 in turbulent flow by an index “T”, their correlation is expressed as[‡]:

$$\frac{u_{\text{term,T}}}{u_{\text{term,0}}} = 1 - 1.18St^{0.7} \exp(-0.47St). \quad (17)$$

[‡] Note that a different definition of the turbulence integral timescale is used here (see section 3.3.1), hence the resulting constants in Eq. (17) are different from those given in Lane et al. (2005).

422 This translates to

$$\frac{C_{D,T}}{C_{D,0}} = \left(\frac{u_{\text{term},T}}{u_{\text{term},0}} \right)^{-2} \quad (18)$$

423 for a corresponding modification factor of the drag coefficient. The Stokes number St gives the
424 ratio of the particle relaxation time in a stagnant fluid $\tau_s = 4d_p(\rho_s/\rho_L + C_{VM})/(3C_{D,0}u_{\text{term},0})$ to
425 the turbulence timescale. For the latter, Lane et al. (2005) simply took the Lagrangian integral
426 timescale following the fluid motion T_L^L .

427 The correlation of Lane et al. (2005), Eq. (17), is based on a summary of data from both experiment
428 and simulation available at that time. Since all of these data were taken at rather low values $St <$
429 1 , a form of the functional dependence was imposed, which ensured that the still-fluid values are
430 approached for both the limits of low and high Stokes numbers in accordance with the general
431 expectation (Good et al., 2014). It is noteworthy that data for both solid particles and gas bubbles
432 are represented in a unified manner by Eq. (17). This can be understood by a mechanism proposed
433 by Spelt and Biesheuvel (1997), which is based on the lift force acting on the particle or bubble.
434 Assuming small enough particle / bubble size such that the lift coefficient is positive (which is the
435 case for all available data and also for the present applications), they argued for bubbles that the
436 lift force acts to move them preferentially to regions where the turbulent fluctuation velocity is
437 directed downwards. On average this leads to a lower rise velocity, which can be modeled by an
438 increased drag coefficient. Particles will in contrast be moved preferentially to regions where the
439 turbulent fluctuation velocity is directed upwards. But this leads again to a lower settling velocity
440 and hence can also be modeled by an increased drag coefficient.

441 The fact that the Lagrangian integral timescale following the fluid motion T_L^L was used to define
442 the Stokes number, rather than particle-turbulence interaction timescale T_L^S has led us to re-
443 evaluate the model of Lane et al. (2005), Eq. (17). Considering that the drag modification results
444 from an interaction between particles and turbulence, use of the latter seems more appropriate.
445 Moreover, T_L^S takes into account the crossing trajectories effect. Due to the appearance of the ratio
446 $k^{1/2}/u_{\text{rel}}$ (see Eq. (13a, b)), this at least in principle captures also the dependence of the drag
447 modification on this second parameter, which is well-known from experimental and simulation
448 studies (Spelt and Biesheuvel, 1997; Poorte and Biesheuvel, 2002). A more recent experimental
449 investigation by Doroodchi et al. (2008) showed that when the particle size becomes comparable
450 to the turbulent lengthscale, the parameter d_p/Λ has an effect, too. If this parameter is small, the
451 drag modification is expected to become independent thereof. Our re-evaluation includes the data
452 from Doroodchi et al. (2008) as well as simulation data from Mazzitelli et al. (2003) in addition to
453 the data from Spelt and Biesheuvel (1997), Brucato et al. (1998b), and Poorte and Biesheuvel
454 (2002), on which the original proposal of Lane et al. (2005) was based. The results are shown in
455 Figure 1.

456 Figure 1 (a) employs the original definition of the Stokes number based on T_L^L as in Lane et al.
457 (2005). The doubly logarithmic scaling makes the deviations between the correlation and the data
458 more readily apparent, but also the deviations between different datasets. The additional data from
459 Mazzitelli et al. (2003) blend quite well with the originally used ones, while the data from
460 Doroodchi et al. (2008) are rather distinct and poorly represented by the correlation Eq. (17) with
461 this definition of the Stokes number.

462 In Figure 1 (b) the presently proposed definition of the Stokes number in terms of T_L^S is used. At
 463 lower values of $St < 1$ most of the data now show a somewhat more coherent trend. There is one
 464 exceptional dataset from Spelt and Biesheuvel (1997, red squares with crosses), which now exhibits
 465 a distinct behavior. This dataset differs from all others by a rather high value of the parameter d_p/Λ
 466 as shown in the legend. As discussed above, under this circumstance a different behavior could be
 467 expected. The data from Doroodchi et al. (2008) now appear at significantly higher values of the
 468 Stokes number and much more in line with the trend suggested by the functional form of Eq. (17)
 469 (but now with a different definition of St). Since these data are also taken at rather high values of
 470 d_p/Λ , however, this agreement may just be fortunate. A last noteworthy observation is that another
 471 one of the datasets from Spelt and Biesheuvel (1997, green empty squares), which appeared at a
 472 single value of St in Figure 1 (a) now is spread over a range of values.

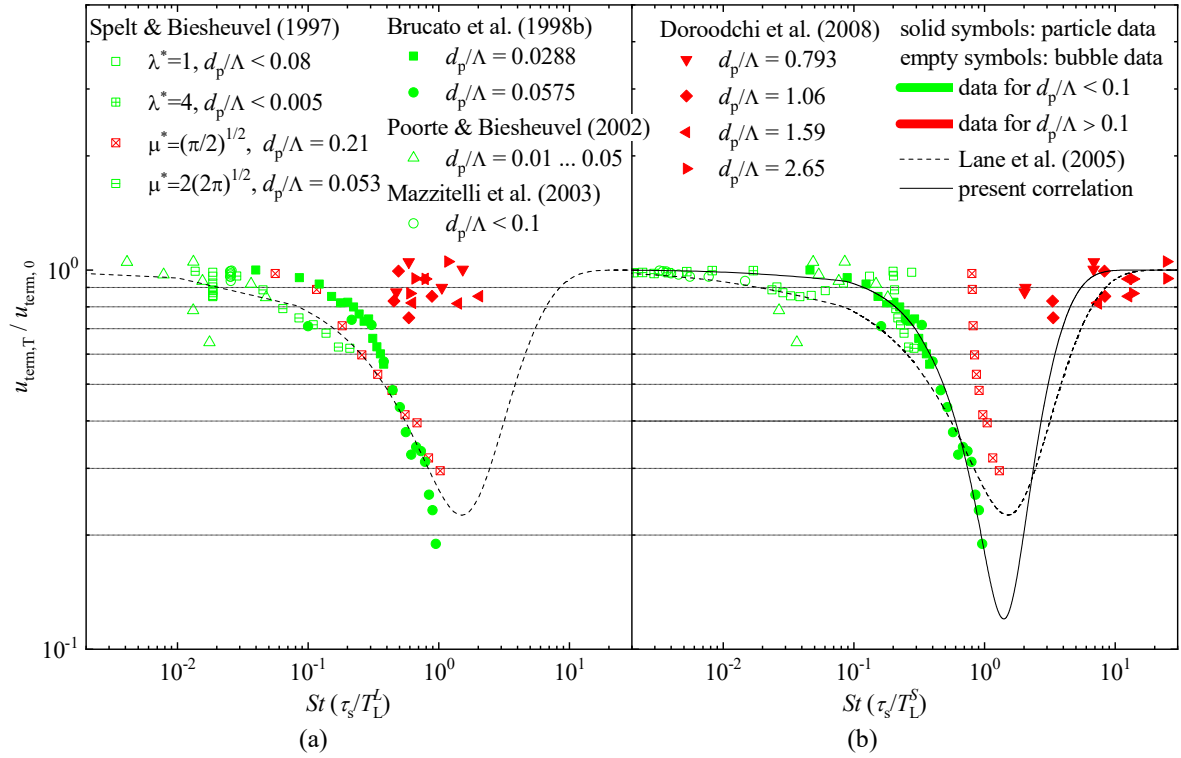
473 Comparing Figure 1 (a) and (b), it may be stated that at least the same quality of agreement is
 474 possible by basing the definition of the Stokes number on T_L^S rather than on T_L^L . This takes into
 475 account the crossing trajectories effect and provides a possibility to include the dependence on a
 476 second relevant parameter, namely $k^{1/2}/u_{rel}$, in addition to τ_s/T_L^L . Moreover this definition is
 477 commonly used in models of turbulent dispersion (e.g. de Bertodano (1998), see section 3.3.3) so
 478 that a unified description of both aspects is obtained. Some reservation has to be made, that cases
 479 with $d_p/\Lambda > 0.1$ may require a more elaborated model accounting for the dependency of the drag
 480 modification also on this third parameter.

481 For a quantitative model, we keep the functional form suggested by Lane et al. (2005) and fit the
 482 parameter values to the data of Figure 1 (b). Data with $d_p/\Lambda > 0.1$, for which this form may not
 483 be adequate (symbols colored in red), have been excluded from the fit. The re-evaluated correlation
 484 becomes

$$\frac{u_{term,T}}{u_{term,0}} = 1 - 2.23St^{1.4}\exp(-St), \quad (19)$$

485 where the Stokes number is defined as $St = \tau_s/T_L^S$. It is shown as the solid line in Figure 1 (b).
 486 The steep decrease of the drag modification factor in the range $0.1 < St \leq 1$ is captured much
 487 better by the revised correlation Eq. (19) than by just changing the definition of St in Eq. (17),
 488 which is shown as the dashed line in Figure 1 (b). Upon further increasing St both correlations
 489 reach a minimum at $St \approx 1.5$, where unfortunately insufficient data are available to precisely judge
 490 the lowest occurring value. Beyond $St \approx 10$ both correlations approach unity. The agreement with
 491 the data of Doroodchi et al. (2008) at higher St is surprising as these were not included in fitting
 492 the correlations.

493 Based on these findings, the presently proposed correlation, Eq. (19), is applied to model the effect
 494 of turbulence on the drag force as it represents the best currently available description, although
 495 there remains an obvious need for further research to fill the mentioned gaps in understanding.



496
497

498 Figure 1. Predictions for the drag modification factor $u_{\text{term},T}/u_{\text{term},0}$ according to the presently proposed correlation,
499 Eq. (19), and the earlier one from Lane et al. (2005), Eq. (17), (represented by solid and dashed lines, respectively) for
500 $10^{-3} < St \leq 30$ compared with experimental and simulation data. Solid and empty symbols denote particle and
501 bubble data, respectively. Green and red colors denote data for $d_p/\Lambda < 0.1$ and $d_p/\Lambda > 0.1$, respectively. The Stokes
502 number is defined as $St = \tau_s/T_L^L$, i.e. as in Lane et al. (2005), in part (a) and as $St = \tau_s/T_L^S$, i.e. as proposed here, in
503 part (b).

504 3.3.3 Turbulent dispersion

505 Turbulent dispersion is significant when the size of the turbulent eddies is larger than the particle
506 size. In stirred tank flows, as indicated in section 2.3, the particle size ($0.1 \text{ mm} \leq d_p \leq 1 \text{ mm}$) is
507 larger than the Kolmogorov lengthscale ($O(10^{-2}) \text{ mm}$) but at least an order of magnitude smaller
508 than that of the energy-containing eddies ($O(10^1) \text{ mm}$). As a result, turbulent dispersion will be
509 significant.

510 A rational way to study turbulent dispersion in turbulence is the probability density function (PDF)
511 approach, which is based on a phase-space formulation of the particle equation of motion including
512 turbulent fluctuations. A comprehensive review of different dispersion models obtained using this
513 approach can be found in Reeks, Simonin, and Fede (2017). For simulations concerning solid
514 dispersion in stirred tank flows, a frequently used model is the one proposed by Reeks (1991).
515 Following de Bertodano (1998) the resulting turbulent dispersion force takes the form[§]

[§] The original version of this correlation (Reeks, 1991) is derived based on the assumption of low Re_p (so that the Stokes drag obeys), while later de Bertodano (1998) found it applicable to describe turbulent dispersion also for conditions with moderate Re_p .

$$\mathbf{F}_S^{\text{disp}} = -C_{D,0} \frac{1}{2} d_p^{-1} \rho_L u_{\text{rel}} \frac{1}{1 + St} T_L^S k \nabla \alpha_S. \quad (20)$$

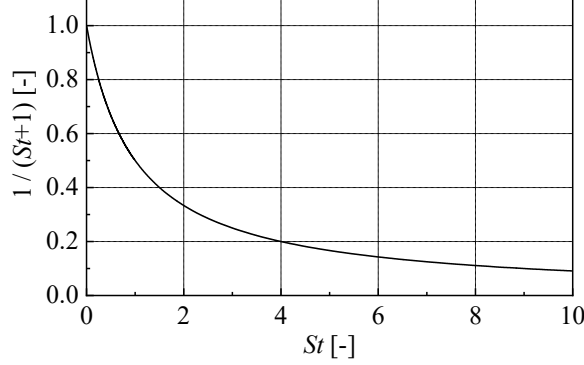


Figure 2. Variation of the magnitude of the turbulent dispersion force.

516
517

518 For later reference, the variation of the magnitude of the turbulent dispersion force with increasing
519 Stokes number is illustrated in Figure 2.

520 An alternative approach is to apply Reynolds-decomposition and -averaging also to the modeled
521 drag force. The most frequently used model following this approach is the Favre averaged drag
522 (FAD) model proposed by Burns et al. (2004), where a similar form as Eq. (20) but with $St = 0$ is
523 obtained. For non-inertial particles, i.e. in the limit $St \rightarrow 0$, the turbulent dispersion forces obtained
524 by the PDF and FAD approaches agree. However, the effect of particle inertia is not accounted for
525 by the FAD approach, which thus would predict too strong dispersion for inertial particles.

526 3.4 SSG Reynolds stress model

527 Only the turbulence in the continuous phase is considered, i.e. the dispersed phase model is applied.
528 The index 'L' is then dropped throughout this section for notational convenience. The transport
529 equation for the Reynolds stress tensor $\mathbf{R} = \langle \mathbf{u}' \otimes \mathbf{u}' \rangle$ is given as

$$\begin{aligned} \frac{\partial}{\partial t} (\alpha \rho \mathbf{R}) + \nabla \cdot (\alpha \rho \mathbf{u} \otimes \mathbf{R}) &= \nabla \cdot (\alpha (\boldsymbol{\mu}^{\text{mol}} + C_s \boldsymbol{\mu}^{\text{turb}}) \nabla \otimes \mathbf{R}) \\ &+ \alpha \rho \left(\mathbf{P} + \boldsymbol{\Phi} - \frac{2}{3} \varepsilon \mathbf{I} + \mathbf{G} \right), \end{aligned} \quad (21)$$

530 and that for the isotropic turbulent dissipation rate ε as

$$\begin{aligned} \frac{\partial}{\partial t} (\alpha \rho \varepsilon) + \nabla \cdot (\alpha \rho \mathbf{u} \varepsilon) &= \nabla \cdot (\alpha (\boldsymbol{\mu}^{\text{mol}} + C_\varepsilon \boldsymbol{\mu}^{\text{turb}}) \cdot \nabla \varepsilon) \\ &+ \alpha \rho \frac{\varepsilon}{k} \left(C_{\varepsilon,1} \frac{1}{2} \text{tr}(\mathbf{P}) - C_{\varepsilon,2} \varepsilon \right). \end{aligned} \quad (22)$$

531 Individual terms appearing on the right side of equation (21) describe diffusion, production,
532 pressure-strain correlation, dissipation, and generation due to body forces (here frame rotation).

533 Compared with isotropic two-equation turbulence models (like for instance the $k - \omega$ SST model),
534 the diffusion term here involves tensorial viscosities:

$$\boldsymbol{\mu}^{\text{mol}} = \mu^{\text{mol}} \mathbf{I}, \quad \boldsymbol{\mu}^{\text{turb}} = \frac{\rho k}{\varepsilon} \mathbf{R}, \quad (23)$$

535 the latter of which is anisotropic. The production term \mathbf{P} is evaluated exactly in terms of the
536 velocity gradient $\nabla \mathbf{u}$ and \mathbf{R} , and its component notation reads

$$P_{ij} = - \left(\frac{\partial u_i}{\partial x_k} R_{jk} + \frac{\partial u_j}{\partial x_k} R_{ik} \right). \quad (24)$$

537 The generation term \mathbf{G} due to frame rotation is given in component notation as

$$G_{ij} = 2\mu^{\text{mol}} \Omega_k (D_{im} \varepsilon_{jkm} + D_{jm} \varepsilon_{ikm}), \quad (25)$$

538 where \mathbf{D} is the strain rate tensor, $\boldsymbol{\Omega}$ the frame angular velocity, and ε_{ijk} is the Levi-Chivita factor
539 defined by

$$\varepsilon_{ijk} = \begin{cases} 1, & \text{if } (i, j, k) \text{ are cyclic,} \\ -1, & \text{if } (i, j, k) \text{ are anticyclic,} \\ 0, & \text{otherwise.} \end{cases} \quad (26)$$

540 Since $\text{tr}(\mathbf{G}) = 0$ from the definition Eq. (25) it does not appear in the equation for the turbulent
541 dissipation rate, Eq. (22).

542 Considerable attention has been devoted to the modeling of the pressure-strain correlation $\boldsymbol{\phi}$ due
543 to its crucial role in redistributing the Reynolds stress components. According to Speziale, Sarkar,
544 and Gatski (1991) this term is given in component notation as

545

546

$$\begin{aligned} \phi_{ij} = & - \left[C_{1a} \varepsilon + C_{1b} \frac{1}{2} \text{tr}(\mathbf{P}) \right] A_{ij} + C_2 \varepsilon \left[A_{ik} A_{kj} - \frac{1}{3} A_{mn} A_{mn} \delta_{ij} \right] \\ & + \left[C_{3a} - C_{3b} (A_{ij} A_{ij})^{\frac{1}{2}} \right] k D_{ij} + C_4 k \left[A_{ik} D_{jk} + A_{jk} D_{ik} - \frac{2}{3} A_{mn} D_{mn} \delta_{ij} \right] \\ & + C_5 k (A_{ik} W_{jk} + A_{jk} W_{ik}), \end{aligned} \quad (27)$$

547 where \mathbf{A} and \mathbf{W} denote the anisotropy and rotation rate tensors, respectively, with components

$$A_{ij} = \frac{R_{ij}}{2k} - \frac{1}{3} \delta_{ij}, \quad W_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} - \frac{\partial u_j}{\partial x_i} \right) + \varepsilon_{ijk} \cdot \Omega_k. \quad (28)$$

548 For the coefficients appearing in the equations above, the standard values of ANSYS CFX (ANSYS
549 2018) have been used, which are summarized in Table 5.

550

Table 5: Coefficient values for the SSG RSM.

ε -equation	C_ε	$C_{\varepsilon 1}$	$C_{\varepsilon 2}$					
		0.18	1.45	1.83				
\mathbf{R} -equations	C_s	C_{1a}	C_{1b}	C_2	C_{3a}	C_{3b}	C_4	C_5
	0.22	3.40	1.80	4.20	0.80	1.30	1.25	0.40

551 4 DESCRIPTION OF THE SIMULATIONS

552 4.1 Investigated tests

553 The data used for validation should contain information on the volume fraction and average
554 velocities so that the modeling of the particle forces may be validated independently. Data relating
555 to fluctuating velocities such as TKE or Reynolds stresses are needed in order to judge the quality
556 of the turbulence model. According to the literature overview of section 2.2, the following datasets
557 were selected to provide a comprehensive validation database that meets the criteria above: Nouri
558 and Whitelaw (1992), Guha et al. (2007), Montante et al. (2012), and Tamburini et al. (2013). In
559 addition, the LES simulation from Guha et al. (2008) is considered as well, since it provides highly
560 resolved simulation results matching the experiment of Guha et al. (2007). For most experiments,
561 a standard tank configuration (Shi and Rzehak, 2018) was used. The solid fraction considered was
562 at most 1% in all experiments so as to satisfy a dilute suspension condition. All selected cases
563 correspond to complete suspension conditions. Other experimental details are summarized in Table
564 6.

565 Nouri and Whitelaw (1992) conducted both single and two-phase flow studies in a 294 mm
566 diameter tank with an impeller rotation speed of 313 rpm. Measurements were performed in a
567 vertical plane placed mid-way between two baffles, and only solid phase information was provided
568 in the two-phase flow measurement. Radial profiles of mean velocities in axial and tangential
569 directions and fluctuation velocity in the axial direction were measured at three horizontal positions
570 of $z/H = 0.068, 0.510, \text{ and } 0.782$. In the near impeller region, axial profiles of mean velocities in
571 radial and axial directions were measured at two axial positions of $2r/D_t = 0.347 \text{ and } 0.463$ in the
572 range of $-1.5 \leq 2z_{\text{bla}}/H_{\text{bla}} \leq 1.5$ (with z_{bla} denoting the axial coordinate with the origin at the impeller
573 disk). An axial profile of local solid fraction was measured at the radial position of $2r/D_t = 0.136$.

574 Also both single and two-phase flow were investigated by Montante et al. (2012) with a 232 mm
575 diameter tank and an impeller rotation speed of 852 rpm. Measurements were performed in a
576 vertical plane in between $z/H = 0.2$ and $z/H = 0.6$ placed mid-way between two baffles. Axial
577 profiles of radial and axial components of both mean and fluctuating liquid velocities were
578 measured at $2r/D_t = 0.88$ and 0.96 .

579 A two-phase flow system was studied by Guha et al. (2007) with a 200 mm diameter tank and an
580 impeller rotation speed of 1000 rpm. Measurements were conducted via the CARPT technique and
581 only azimuthally averaged quantities were provided. Radial profiles of mean solid velocities in
582 radial, axial, and tangential directions were measured at three horizontal positions of $z/H = 0.075,$
583 $0.34,$ and 0.65 . The corresponding LES simulation of Guha et al. (2008) additionally provides a
584 radial profile of local solid fraction at a horizontal position of $z/H = 0.34$.

585 Another two-phase flow system was investigated by Tamburini et al. (2013) with a 190 mm
586 diameter tank and impeller rotation speeds ranging from 300 to 600 rpm. Differing from the
587 standard configuration, the tank here was un-baffled and the turbine diameter was half that of the
588 tank diameter. Measurements were performed in a vertical plane placed mid-way between two
589 baffles, where in contrast to the previous works radially averaged axial profiles of solid fraction
590 are provided.

591

592

Table 6: Parameters for the investigated test cases.

Reference	D_t (mm)	C_i (mm)	H_{bla} (mm)	Ω (rpm)	u_{tip} (m s ⁻¹)	ρ_S/ρ_L (-)	$\bar{\alpha}_{S,ave}$ (%)	d_p (mm)	Available data
Nouri and Whitelaw (1992)	294	73.5	19.6	313	1.61	-	-	-	$\bar{u}_r, \bar{u}_z, \bar{u}_\theta; u'_z$
Montante et al. (2012)	232	77.3	15.5	852	3.45	-	-	-	$\bar{u}_r, \bar{u}_z; u'_r, u'_z$
Nouri and Whitelaw (1992)	294	73.5	19.6	313	1.61	1.32	0.50	0.665	$\bar{u}_{S,r}, \bar{u}_{S,z}, \bar{u}_{S,\theta}; \bar{\alpha}_S$
Guha et al. (2007, 2008)	200	66.7	13.3	1000	3.49	2.50	1.00	0.300	$\bar{u}_{S,r}, \bar{u}_{S,z}, \bar{u}_{S,\theta}; \bar{\alpha}_S$ (LES)
Montante et al. (2012)	232	77.3	15.5	852	3.45	2.47	0.05	0.115	$\bar{u}_{L,r}, \bar{u}_{L,z}; u'_{L,r}, u'_{L,z}$
							0.05	0.775	$\bar{u}_{L,r}, \bar{u}_{L,z}; u'_{L,r}, u'_{L,z}$
							0.15	0.775	$\bar{u}_{L,r}, \bar{u}_{L,z}; u'_{L,r}, u'_{L,z}$
Tamburini et al. (2013)	190	63.3	19.0	300	1.49	2.48	0.0081	0.138	$\bar{\alpha}_S$
				600	2.98	2.48	0.0081	0.138	$\bar{\alpha}_S$

594

595 4.2 Solution domain, boundary conditions and numerical approach

596 ANSYS CFX release 19.2 is used for the numerical simulations. This software solves the three-
597 dimensional unsteady Reynolds-averaged Navier-Stokes equations with a control volume based
598 finite-element discretization. For the problem considered, the advection terms are discretized using
599 the high resolution scheme proposed in Barth and Jespersen (1989), while the solution is advanced
600 in time with a second order backward Euler scheme. Other details regarding the discretization of
601 the diffusion and pressure gradient terms as well as the solution strategy are detailed in ANSYS
602 Inc. (2018, section 11).

603 The simplification of the computational domain, the arrangement concerning the position of the
604 baffles and the impellers, and the implementation of the mixing-plane model of the MRF method
605 (ANSYS Inc., 2018) to couple the results from the rotating and the static blocks can be found in
606 Shi and Rzehak (2018). A still and homogeneous suspension is taken as the initial condition. On
607 the walls no-slip and free slip conditions are applied for the liquid and solid phases, respectively,
608 while the scalable wall function is used to specify the wall boundary condition in SSG RSM
609 turbulence model. At the top of the suspension, a free slip wall is introduced.

610 Fully structured meshes are used for all investigated cases (see Table 7 for the mesh details) with
611 comparable average spacings in radial, azimuthal, and axial direction as those of Shi and Rzehak
612 (2018), where geometries of similar dimensions were investigated. To avoid numerical difficulties,
613 for each case, the calculation is performed at first in pseudo-transient mode for 50 rotations and
614 then switched to transient mode for 20 rotations. The time step for each stage is set again in
615 accordance with Shi and Rzehak (2018) such that a rotation of 4° per time step results in order to
616 achieve low residuals ($\leq 10^{-5}$). Averaged results are obtained during the last 10 rotations. Following
617 these numerical settings, the adequacy of the resulting solutions was established in Shi and Rzehak
618 (2018), in which test cases with impeller rotation speed up to 450 rpm were considered. Since a
619 higher impeller rotation speed is involved in some of the investigated cases, a further grid
620 independency study is presented in Appendix A, where results for the mean and fluctuation
621 velocities are discussed for the single-phase flow case of Guha et al. (2007) with an impeller
622 rotation speed of 1000 rpm.

623

624

Table 7: Parameters for meshes for all investigated test cases.

Test case	Tank volume			Impeller blade			Overall	CPU time (with 32 processors)
	N_r	N_θ	N_z	N_r	N_θ	N_z	N_{tot}	
Nouri and Whitelaw (1992)	120	150	150	36	5	36	2.70×10^6	276 h
Guha et al. (2007)	101	120	120	30	4	30	1.45×10^6	130 h
Montante et al. (2012)	95	120	131	22	4	32	1.49×10^6	144 h
Tamburini et al. (2013)	115	99	108	36	3	36	1.23×10^6	100 h

625

626 For unsteady RANS simulations conducted in multiple reference frames, the calculation of
627 averages and fluctuations needs to be carefully considered in order to match the experimentally
628 obtained values. For a detailed discussion, the reader is referred once again to Shi and Rzehak
629 (2018).

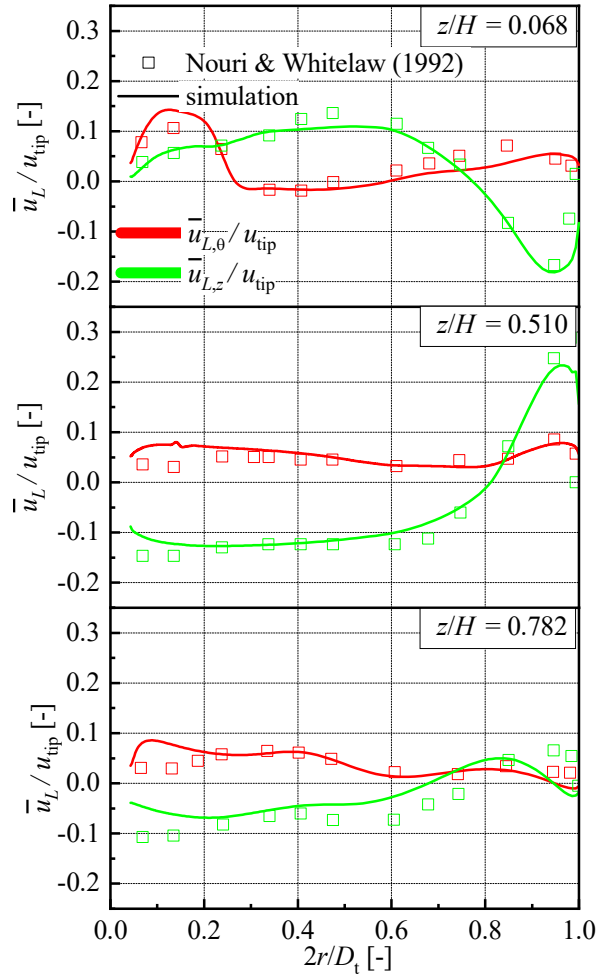
630 5 RESULTS AND DISCUSSION

631 5.1 Single-phase results

632 Single-phase flow simulations are conducted first to get an idea of the performance that can be
633 expected for a RANS turbulence model, namely the SSG RSM. Model assessment is done first for
634 the mean liquid velocity and then for the liquid velocity fluctuations using the data of Nouri and
635 Whitelaw (1992) and Montante et al. (2012). For both, mean and fluctuations, this comprises
636 several profiles along radial and axial directions throughout the entire free flow region in the tank
637 between the impeller and the baffles and all three components of velocity.

638 5.1.1 Mean velocity

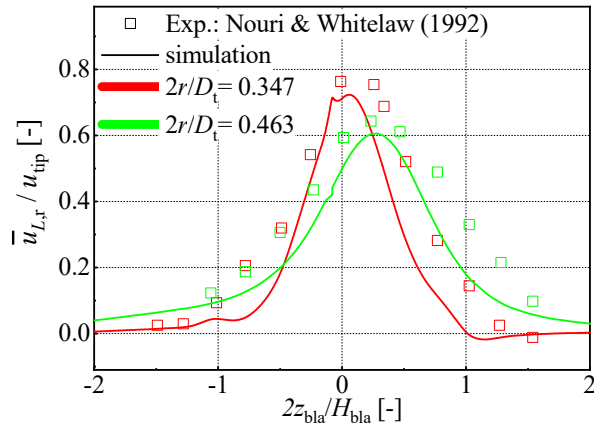
639 Figure 3 compares simulation results for radial profiles of tangential and axial mean liquid velocity
640 with the measured data from Nouri and Whitelaw (1992). At all three heights of $z/H = 0.068$,
641 0.510 , and 0.782 , generally very good agreement with the experimental data is achieved by the
642 current simulation. Along the radial direction, some deviation from the measured data can be
643 observed at $0.05 \leq 2r/D_t \leq 0.15$ (i.e. the region near the tank shaft) and $0.9 \leq 2r/D_t \leq 1$ (i.e.
644 the region near the tank wall).



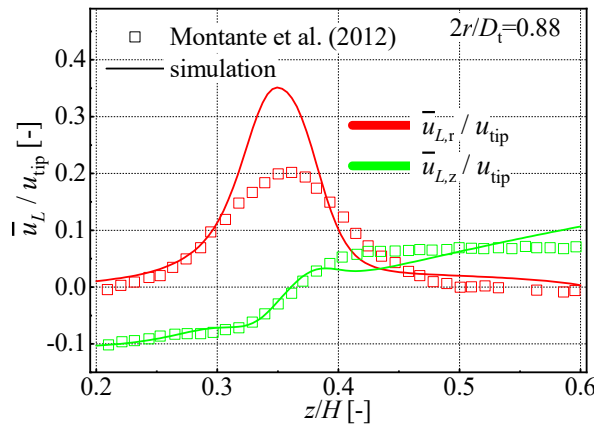
645
 646 Figure 3. Comparison of present simulation results (lines) and measured data (symbols) from Nouri and Whitelaw
 647 (1992) for the tangential (red) and axial (green) components of mean liquid velocity. Radial profiles over the entire
 648 tank radius are shown at different heights as indicated on each panel.

649 Comparisons of axial profiles of mean fluid velocity between the simulations and the measured
 650 data from Nouri and Whitelaw (1992) and Montante et al. (2012) are shown in Figure 4 and Figure
 651 5, respectively. Only data for the radial component of mean liquid velocity are provided by Nouri
 652 and Whitelaw (1992). Also, as shown in Figure 4, only a portion of the tank height around the
 653 impeller has been considered. The predicted peak values at both radial positions are in quantitative
 654 agreement with the ones observed in the experiment, but the predicted profiles show a bit narrower
 655 structures than found in the measured data.

656 The experiment of Montante et al. (2012) provides data for radial and axial components of mean
 657 liquid velocity at the radial position of $2r/D_t = 0.88$, which is close to the tank wall. The predicted
 658 axial component agrees quite well with the measured data, however the peak of the radial
 659 component is significantly overestimated.



660
 661 Figure 4. Comparison of present simulation results (lines) and measured data (symbols) from Nouri and Whitelaw
 662 (1992) for the radial component of mean liquid velocity in the near impeller region at $2r/D_t = 0.347$ (red) and
 663 $2r/D_t = 0.463$ (green). Axial profiles restricted to a height range around the impeller are shown.

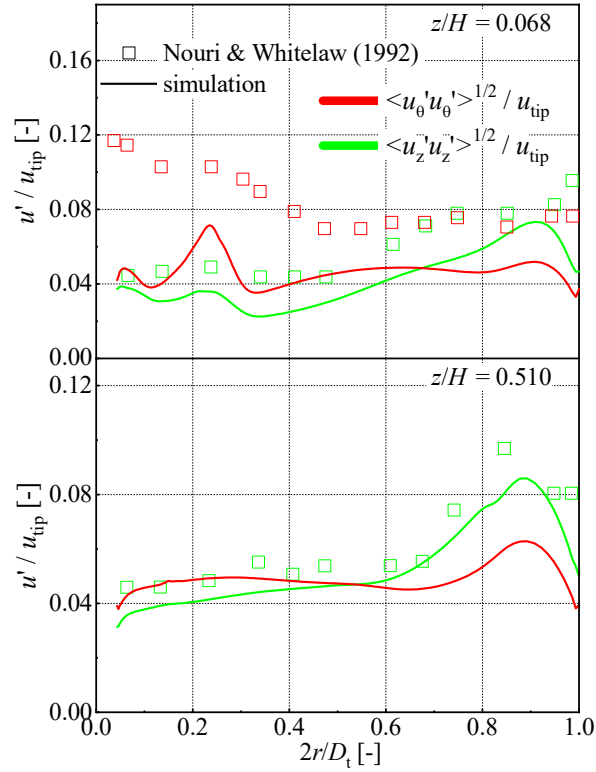


664
 665 Figure 5. Comparison of present simulation results (lines) and measured data (symbols) from Montante et al. (2012)
 666 for the radial (red) and axial (green) components of mean liquid velocity. Axial profiles restricted to a height range of
 667 $0.2 \leq z/H \leq 0.6$ are shown at a radial position of $2r/D_t = 0.88$.

668 In summary, taken together with the single-phase results from Shi and Rzehak (2018), it may be
 669 stated that for the mean velocities good predictions are obtained at lower rotation speeds Ω at least
 670 up to 450 rpm, while deviations occur at higher values certainly from 850 rpm on. Where deviations
 671 occur, the most prominent ones are localized near the impeller blades and less significant ones near
 672 the tank wall and impeller shaft.

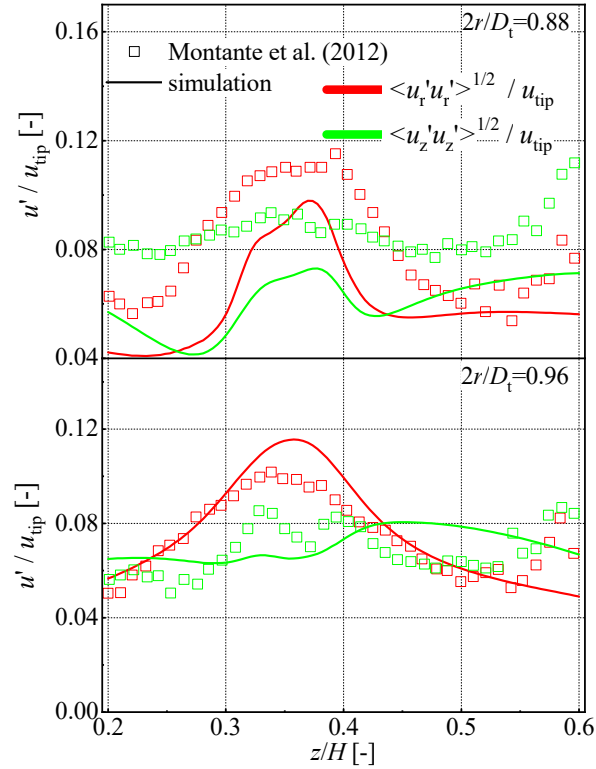
673 5.1.2 Turbulent fluctuations

674 Radial profiles for the fluctuating liquid velocity are provided by Nouri and Whitelaw (1992) at
 675 heights of $z/H = 0.068$ and 0.51 . For the former, both tangential and axial components are
 676 available, while for the latter only the axial component is provided. As seen in Figure 6, the
 677 agreement between simulation and experiment is good for the component at the higher
 678 measurement position. At the lower measurement position, which is quite close to the tank bottom,
 679 a moderate underprediction is seen for the axial component and a larger one for the tangential
 680 component. The proximity of the tank bottom suggests that this deviation might be caused by a
 681 wall-effect, which is a known issue in standard RSMs (Launder and Sandham, 2002, section 2).



682
 683 Figure 6. Comparison of present simulation results (lines) and measured data (symbols) from Nouri and Whitelaw
 684 (1992) for the tangential (red) and axial (green) components of fluctuating liquid velocity. Radial profiles over the
 685 entire tank radius are shown at different heights as indicated on each panel.

686 Figure 7 compares predictions of the axial profiles of the radial and axial components of fluctuating
 687 liquid velocity at the radial positions of $2r/D_t = 0.88$ and 0.96 with the measured data from
 688 Montante et al. (2012). According to the measured data, the radial component is larger than the
 689 axial one in the impeller stream (i.e. for roughly $0.25 < z/H < 0.45$) and becomes smaller than
 690 the latter at regions outside the impeller stream. This qualitative feature is captured by the
 691 predictions while the quantitative agreement is only mediocre. Farther away from the tank wall, at
 692 $2r/D_t = 0.88$, both fluctuation components are significantly underestimated. Nearer to the wall, at
 693 $2r/D_t = 0.96$, deviations are much less severe with both over- and underestimation occurring in
 694 different parts of the profiles.



695
 696 Figure 7. Comparison of present simulation results (lines) and measured data (symbols) from Montante et al. (2012)
 697 for the radial (red) and axial (green) components of fluctuating liquid velocity. Axial profiles restricted to a height
 698 range of $0.2 \leq z/H \leq 0.6$ are shown at different radial positions as indicated on each panel.

699 In summary, again taken together with the single-phase results from Shi and Rzehak (2018), it may
 700 be stated that for the turbulent fluctuations, reasonable predictions are only obtained at very low
 701 rotation speeds Ω smaller than 200 rpm. At larger values of Ω mostly only mediocre agreement
 702 with the measured values is found though qualitative features of the data are reproduced.

703 5.2 Two phase results

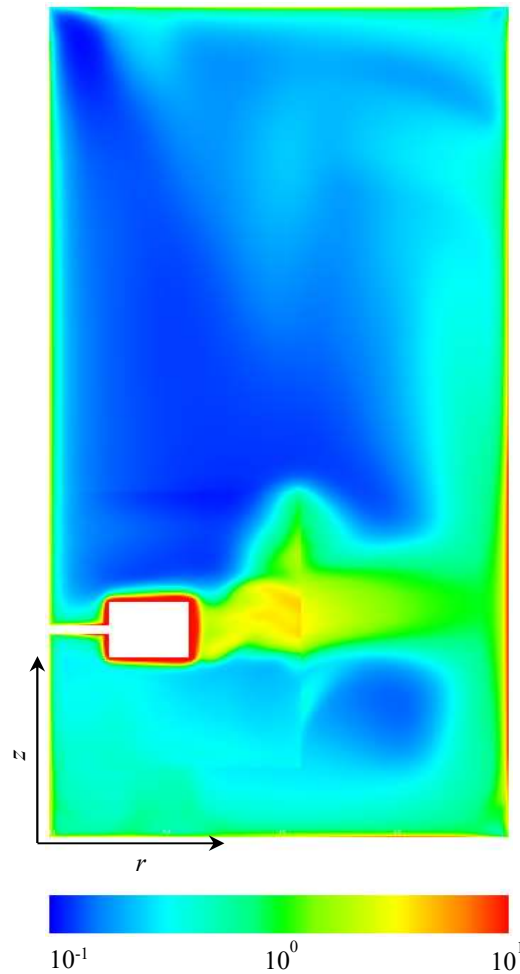
704 The full model presented in section 3 will be taken as a baseline for the investigation of two-phase
 705 flows. In addition, seven reduced model variants, summarized in Table 8, are considered to
 706 highlight the importance of various aspects. Two model variants termed T-0.1 and T-0.5 use
 707 identical particle forces as the baseline model, but adopt different settings of the integral timescale
 708 T_L^L , namely $0.1 k/\varepsilon$ and $0.5 k/\varepsilon$ as opposed to $0.224 k/\varepsilon$ for the baseline model. This choice
 709 potentially affects the turbulent dispersion force as well as the drag modification due to turbulence.
 710 The two model variants, drag-SN and drag-Lane, differ from the baseline model in the drag
 711 correlation. Compared with the baseline model, the former disregards the turbulence effects on the
 712 drag while the latter accounts for these effects by the model from Lane et al. (2005) which neglects
 713 the crossing trajectory effects. The model variant disp-FAD differs from the baseline model in the
 714 turbulent dispersion force correlation. Compared with the baseline model, turbulent dispersion is
 715 accounted for by the FAD model from Burns et al. (2004) which assumes negligible particle inertia
 716 and approaches the baseline model in the limit $St \rightarrow 0$. Effects of the lift and virtual mass forces
 717 are assessed by the model variants lift-off and vm-off, respectively, where one of the forces is
 718 simply turned off from the baseline model.

719 Validation of the baseline model is conducted taking the following approach. The selected two
720 phase flow cases of Nouri and Whitelaw (1992) and Guha et al. (2007, 2008) provide relatively
721 comprehensive data, which comprise experimental or LES data of both mean solid velocity and
722 solid fraction for model validation and are considered first. Simulations applying the baseline
723 model as well as all model variants listed in Table 8 are conducted for the two cases to confirm the
724 advantage of the baseline model over all other model variants. Extension of the validation of the
725 baseline model is then made by comparing the simulation results of the baseline model with the
726 measured data from Montante et al. (2012) and Tamburini et al. (2013). The former experiment
727 provides data for the mean and fluctuating liquid velocities and considers varying particle size and
728 solids loading. The latter provides data for the axial profiles of the solid fraction and considers
729 varying impeller rotation speed.

730 Table 8: Summary of particle force correlations used in the various models applied in the present work.

Model abbreviation	Force correlations			
	Drag	Turb. disp.	Lift	Virt. mass
baseline	Eqs. (18) & (19)	de Bertodano (1998, $T_L^L = 0.224 k/\varepsilon$)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
T-0.1	Eqs. (18) & (19)	de Bertodano (1998, $T_L^L = 0.1 k/\varepsilon$)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
T-0.5	Eqs. (18) & (19)	de Bertodano (1998, $T_L^L = 0.5 k/\varepsilon$)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
drag-SN	Schiller & Naumann (1933)	de Bertodano (1998, $T_L^L = 0.224 k/\varepsilon$)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
drag-Lane	Lane et al. (2005)	de Bertodano (1998, $T_L^L = 0.224 k/\varepsilon$)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
disp-FAD	Eqs. (18) & (19)	Burns et al. (2004)	Shi & Rzehak (2019)	$C_{VM} = 0.5$
lift-off	Eqs. (18) & (19)	de Bertodano (1998, $T_L^L = 0.224 k/\varepsilon$)	-	$C_{VM} = 0.5$
vm-off	Eqs. (18) & (19)	de Bertodano (1998, $T_L^L = 0.224 k/\varepsilon$)	Shi & Rzehak (2019)	-

731
732 The Stokes number $St = \tau_s/T_L^S$ is an important parameter, which crucially affects the intensity of
733 the drag and turbulent dispersion forces and, consequently, influences the resulting flow field.
734 Therefore it is of interest to estimate the range of values that has to be expected. In the absence of
735 theoretical estimates that are applicable to stirred tank flows, values from simulations using the
736 baseline model are used for this purpose. For the case of Nouri and Whitelaw (1992) the calculated
737 range of values is about $0.1 \leq St \leq 10$ as shown in Figure 8. The magnitude of St is relatively
738 low in the bulk region, typically lower than 0.5, but increases dramatically near any no-slip wall.
739 This increase is not surprising since when approaching the wall the turbulent kinetic energy k
740 vanishes while the dissipation rate ε approaches its maximum (e.g. Wilcox, 2006). This results in
741 vanishing values of T_L^L and, consequently, T_L^S . Moderate values of St within roughly $0.5 \leq St \leq 5$
742 appear along the impeller stream possibly due to the relatively higher values of dissipation rate
743 appearing in this region (Sbrizzai et al., 2006). In the other three cases, the distribution of St
744 obtained (not shown) does not differ too much from that in Nouri and Whitelaw (1992) although
745 particles with smaller relaxation time are considered. This is likely due to the higher impeller
746 rotation speed involved which causes a more turbulent flow with a smaller value of T_L^L .



747
 748 Figure 8. St calculated from the baseline model in the plane midway between two baffles for the case of Nouri and
 749 Whitelaw (1992). Results for $r \leq 0.57$ and $0.1 \leq h \leq 0.4$ are obtained by averaging the transient results in a frame
 750 rotating with the impeller while for the rest domain time averaged results are obtained in a laboratory frame.

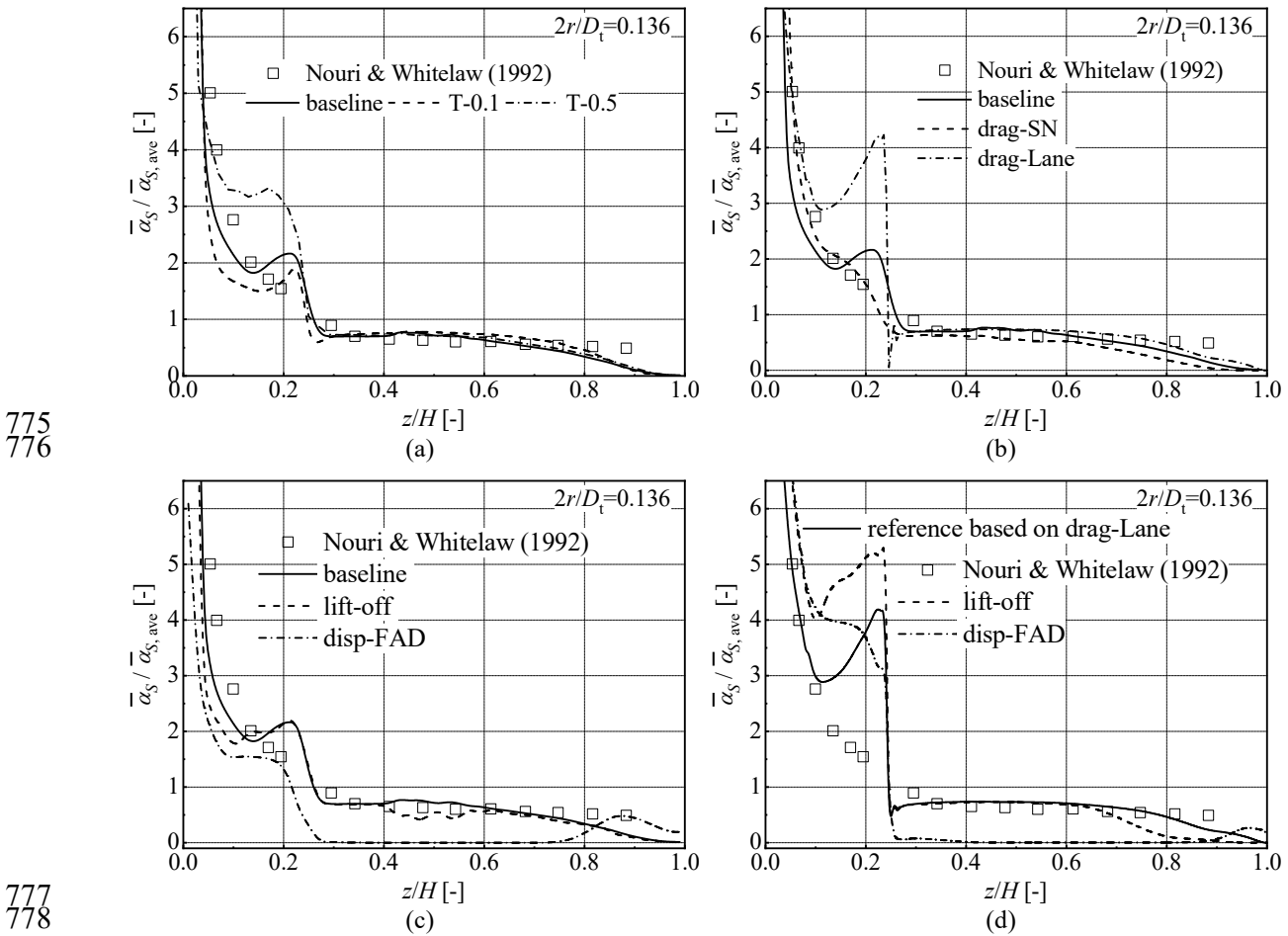
751 5.2.1 Tests from Nouri and Whitelaw (1992)

752 Figure 9 (a) compares the predictions according to the model variants adopting different settings
 753 of the integral timescale T_L^L to the experimental data from Nouri and Whitelaw (1992) for the axial
 754 profile of the solid fraction at the radial position of $2r/D_t = 0.136$. The effect of different settings
 755 is pronounced in the region below the impeller disk, namely for $z/H \leq 0.25$. Simulation results
 756 from the model variants T-0.1 and T-0.5 suffer, respectively, under- and overestimation compared
 757 to the experimental data, while good agreement is obtained by the baseline model. According to
 758 Figure 8 the typical Stokes number range in this region is from 0.5 to 1, within which the drag force
 759 is quite sensitive to the change in St (see Figure 1).

760 A similar comparison concerning the model variants adopting different drag correlations is shown
 761 in Figure 9 (b). When turbulence effects are taken into account the predicted profile shows a peak
 762 below the impeller disk, which is more pronounced with the variant drag-Lane than with the
 763 baseline model. For the model variant drag-SN, which neglects turbulence effects, this peak is
 764 absent. In quantitative terms, the baseline model comes much closer to the experimental data than

765 the variant drag-Lane. The variant drag-SN here performs also very good, but for a more precise
 766 judgement experimental data in the vicinity of the impeller disk are unfortunately lacking. Above
 767 the impeller disk only a slight difference between different model variants can be observed. Note
 768 that the Stokes number in this region according to Figure 8 is around 10^{-1} , based on which the drag
 769 modifications according to Lane et al. (2005) and to the present proposal are both very small.

770 Effects of different models concerning the non-drag forces are illustrated in Figure 9 (c). Compared
 771 with the baseline prediction, the variant disp-FAD gives much lower solid fraction especially above
 772 the impeller. Since the profile is taken quite close to the impeller shaft, where the Stokes number
 773 ranges between 1 and 10 such a significant effect on the turbulent dispersion force may be expected
 774 (see Figure 2). Neglecting the lift force on the other hand does not cause any big changes.

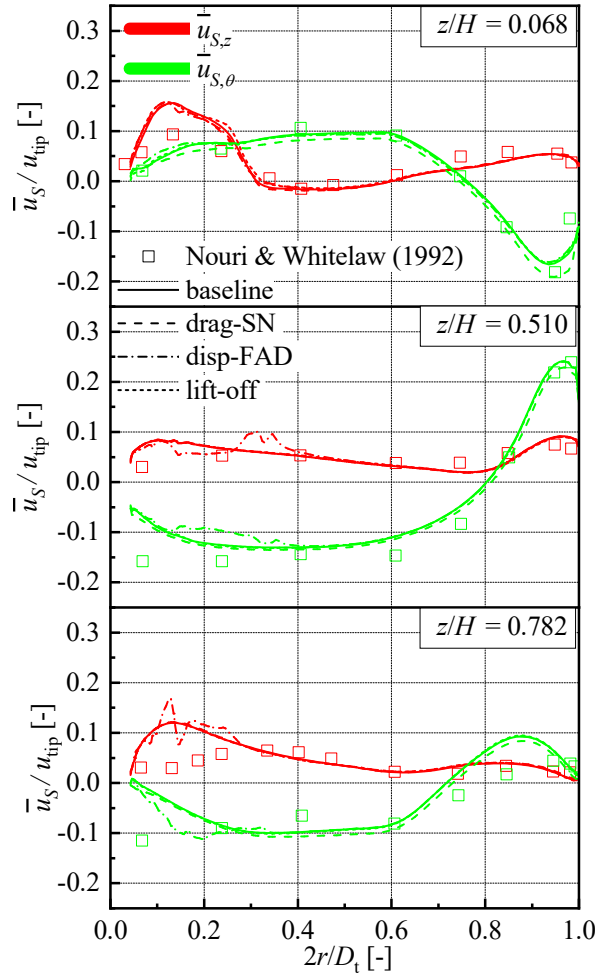


777
 778
 779 Figure 9. Comparison of the simulation results (lines) according to the baseline model and different model variants
 780 indicated in Table 8 and measured data (symbols) from Nouri and Whitelaw (1992) for the solid fraction at $2r/D_t =$
 781 0.136 . Axial profiles over the entire tank height are shown.

782 However, these features are highly interdependent with other parts of the model. As seen in Figure
 783 9 (d), when taking the variant drag-Lane as the reference model, both switching the turbulent
 784 dispersion model from the one proposed by de Bertodano (1998) to the FAD model or turning off
 785 the lift force results in a significant increase in the predicted solid fraction below the impeller disk,
 786 i.e. for $z/H \leq 0.2$. These features are not surprising, as closures for the lift and turbulent dispersion

787 forces used in our simulation depend on the particle-fluid relative velocity, which is essentially
 788 affected by the drag law.

789 The predicted profiles of the model variant vm-off with either the baseline or the drag-Lane model
 790 taken as a reference (not shown in Figure 9 (c) and (d)) reveal hardly any difference from those
 791 using the reference models, which indicates a negligible effect of the virtual mass force here.



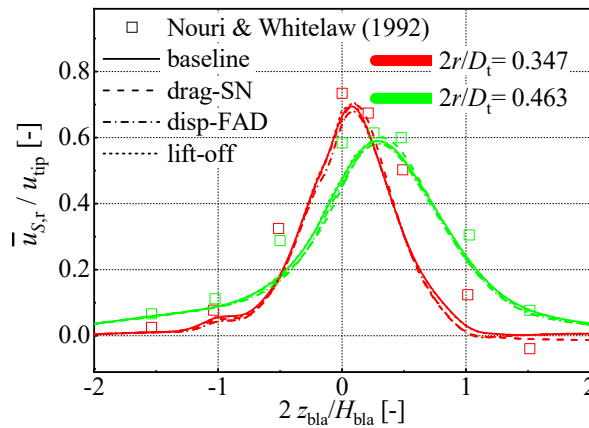
792
 793 Figure 10. Comparison of the simulation results (lines) according to the baseline model and different model variants
 794 listed in Table 8 and measured data (symbols) from Nouri and Whitelaw (1992) for the axial (red) and tangential
 795 (green) components of mean solid velocity. Radial profiles over the entire tank radius are shown at different heights
 796 as indicated on each panel.

797 Figure 10 compares the model predictions to the measured data for the tangential and axial mean
 798 solid velocities along radial profiles at three different heights. Predictions according to some of the
 799 model variants, namely T-0.1, T-0.5, drag-Lane, and vm-off, are omitted as they show hardly any
 800 difference to that of the baseline model (represented by solid lines in Figure 10). Switching to the
 801 variant disp-FAD (represented by dash-dotted lines in Figure 10) introduces some erratic deviations
 802 in both components of the velocity in the lower half of the tank at the two heights above the
 803 impeller. To make absolutely sure that this observation is not caused by numerical effects, this case
 804 has been re-calculated by decreasing the time step in the transient mode from 4° to 1° per time step

805 and simultaneously increasing the number of rotations used for averaging from 10 to 20 rotations
 806 with no difference in the results. For the variant drag-SN (represented by dashed lines in Figure
 807 10) a slight decrease in velocity is seen throughout. Turning off the lift force in variant lift-off
 808 (represented by short-dashed lines in Figure 10) has hardly any effect. Since concerning the solid
 809 fraction, the effect of the lift force was much stronger when changing the reference to drag-Lane,
 810 this case was considered as well (not shown in the figure). It turns out that this change of reference
 811 model does not affect the results concerning the solid velocity.

812 Compared to the experimental data good agreement is found for the baseline predictions in the bulk
 813 region. In the region near the tank shaft, namely for $0.05 \leq 2r/D_t \leq 0.15$, some deviation from
 814 the experimental data can be observed. This type of deviation appeared also in the single-phase
 815 tests and hence can be considered as a drawback of the RANS turbulence model.

816 A similar comparison concerning the radial component of the mean solid velocity at two radial
 817 positions near the impeller is given in Figure 11. The agreement of the baseline prediction with the
 818 experimental data is quite good. All model variants give almost identical profiles as the baseline
 819 model. As before only a selection of variants is shown in the figure, but the omitted ones have even
 820 smaller difference from the baseline model. The erratic deviation suffered by the variant disp-FAD
 821 in the bulk region does not occur here.



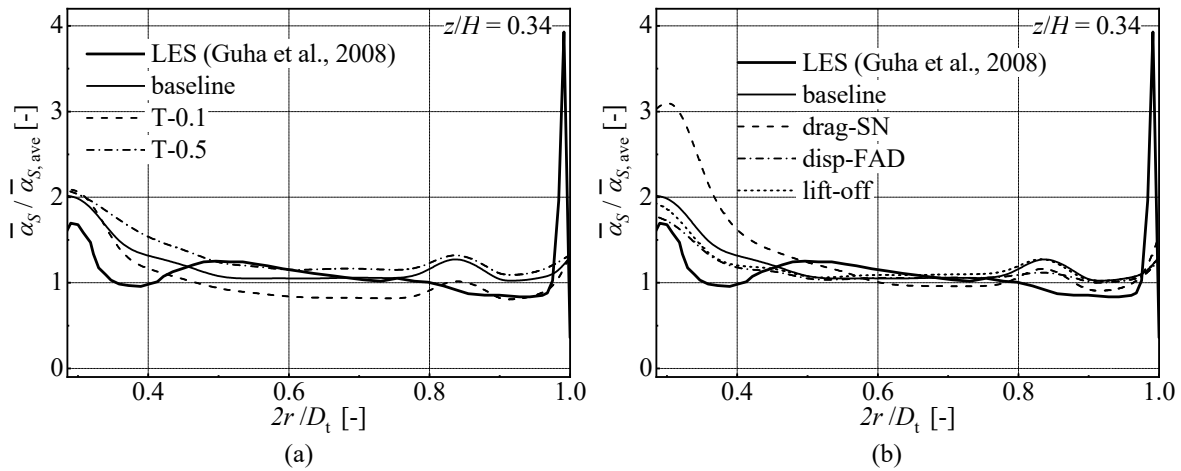
822
 823 Figure 11. Comparison of the simulation results (lines) according to the baseline model and different model variants
 824 listed in Table 8 and measured data (symbols) from Nouri and Whitelaw (1992) for the radial component of mean solid
 825 velocity in the near impeller region. Axial profiles restricted to a height range around the impeller are shown at radial
 826 positions of $2r/D_t = 0.347$ (red) and $2r/D_t = 0.463$ (green).

827 5.2.2 Tests from Guha et al. (2007, 2008)

828 The comparison between the present predictions and the E-L / LES results from Guha et al. (2008)
 829 for the azimuthally averaged radial profile of the solid fraction at the height $z/H = 0.34$ is shown
 830 in Figure 12. A prominent feature of the E-L / LES results is the sharp peak near the wall. The
 831 presence of this peak is possibly a result of particle-wall collisions. Since these are not included in
 832 the baseline model, it is not surprising that such a near-wall peak does not appear in all the current
 833 predictions. Except for the near wall region the agreement of the baseline prediction with the E-L
 834 / LES results is generally acceptable. Results according to the variants T-0.1 and T-0.5 shown in
 835 Figure 12 (a) give, respectively, higher and lower values of solid fraction for $2r/D_t \geq 0.4$.
 836 Approaching $2r/D_t = 0.3$ both predict slightly higher solid fraction. In the region $0.3 \leq 2r/D_t \leq$
 837 0.4 the azimuthally averaged Stokes number (not shown in Figure 8) at the height of the impeller

838 disk has a typical value of $St \approx 1$, which is close to the critical value where according to Figure 1
 839 (b) the strongest drag modification occurs. Departure from this critical value by either an increase
 840 or a decrease in St results in weaker drag modification.

841 Figure 12 (b) illustrates the effects of individual interfacial forces. The variant drag-SN predicts a
 842 significantly higher value of solid fraction compared with the baseline prediction in the region
 843 $0.3 \leq 2r/D_t \leq 0.4$ and deviates strongly from the E-L / LES results. The prediction according to
 844 the variant disp-FAD shows good agreement with the baseline prediction, which is different from
 845 the findings concerning the axial profile of solid fraction in the case of Nouri and Whitelaw (1992).
 846 Predictions according to all other variants, namely drag-Lane (not shown), lift-off, and vm-off (not
 847 shown) show hardly any difference compared with the baseline prediction.

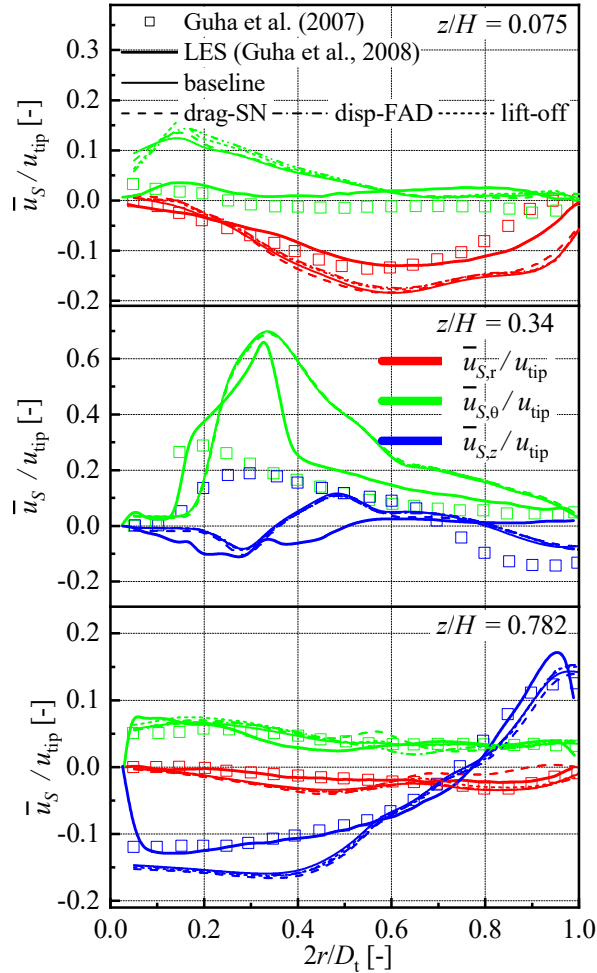


848
 849
 850 Figure 12. Comparison of the simulation results (lines) according to the baseline model and different model variants
 851 listed in Table 8 and the LES result (thick solid lines) from Guha et al. (2008) for azimuthally averaged radial profile
 852 of solid fraction at the height $z/H = 0.34$. Profiles over the radial section outside the impeller disk, i.e. $0.3 \leq 2r/D_t \leq$
 853 1 are shown.

854 Figure 13 compares the present predictions and the previous E-L / LES results for azimuthally
 855 averaged radial profiles of radial, tangential and axial mean solid velocity to the experimental data
 856 from Guha et al. (2007). The previous E-L / LES results show mostly better agreement with the
 857 experimental data than the present baseline prediction. However, at the height $z/H = 0.34$ where
 858 the impeller is located, both approaches fail to provide a reasonable representation of the
 859 experimental data. At the other two heights, namely $z/H = 0.075$ and $z/H = 0.782$, the
 860 agreement between the baseline prediction and the experimental data is generally acceptable except
 861 for the tangential velocity at $z/H = 0.075$, which is obviously overestimated. In a previous E-E /
 862 RANS simulation (Guha et al., 2008, Figure 4 (a)) of this case by adopting the $k - \epsilon$ turbulence
 863 model even the direction of the tangential flow in this region was not captured correctly.

864 Predictions according to all other model variants show only minor differences to the baseline model
 865 as shown for drag-SN, disp-FAD, and lift-off in Figure 13. The variants T-0.1, T-0.5, drag-Lane,
 866 and vm-off are omitted in Figure 13 as they differ even less from the baseline prediction. This
 867 insensitivity of the predictions to various aspects of the interaction between the phases suggests
 868 that the observed deviations from the experimental data may originate from the RANS turbulence
 869 modeling. Also note that the erratic deviation of the variant disp-FAD from the baseline results
 870 found for the test of Nouri and Whitelaw (1992) in the last section does not occur here. This and

871 the corresponding findings concerning the solid fraction profile mentioned above, are possibly due
 872 to impeller rotation speed being much faster here than for the test of Nouri and Whitelaw (1992),
 873 which could cause particle suspension to be dominated by the mixing due to the mean flow such
 874 that turbulent dispersion no longer plays a significant role.

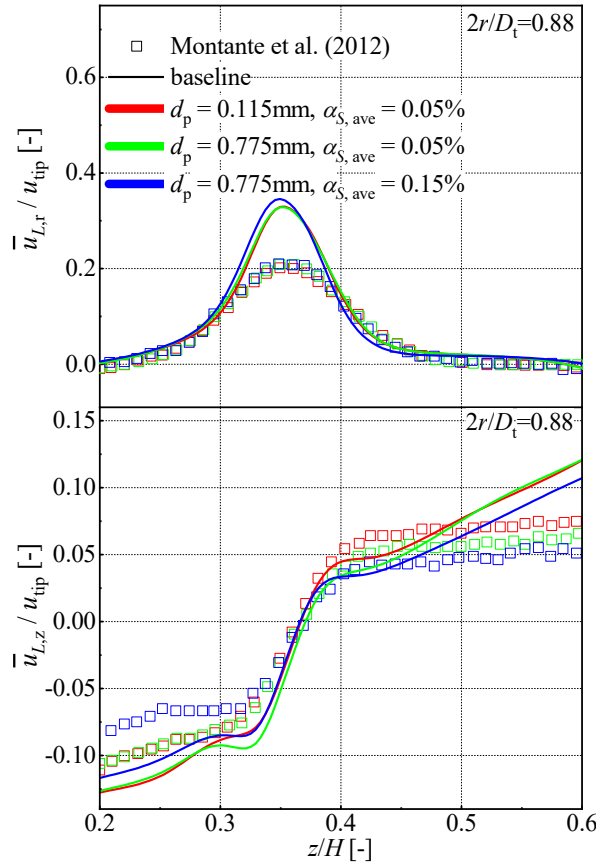


875
 876 Figure 13. Comparison of the simulation results (lines) according to the baseline model and different model variants
 877 listed in Table 8 and the measured data (symbols) from Guha et al. (2007) for azimuthally averaged radial profiles of
 878 the radial (red), tangential (green), and axial (blue) components of mean solid velocity. The E-L / LES results (thick
 879 solid lines) from Guha et al. (2008) are shown for comparison as well. Radial profiles over the entire tank radius are
 880 shown at different heights as indicated on each panel.

881 5.2.3 Tests from Montante et al. (2012)

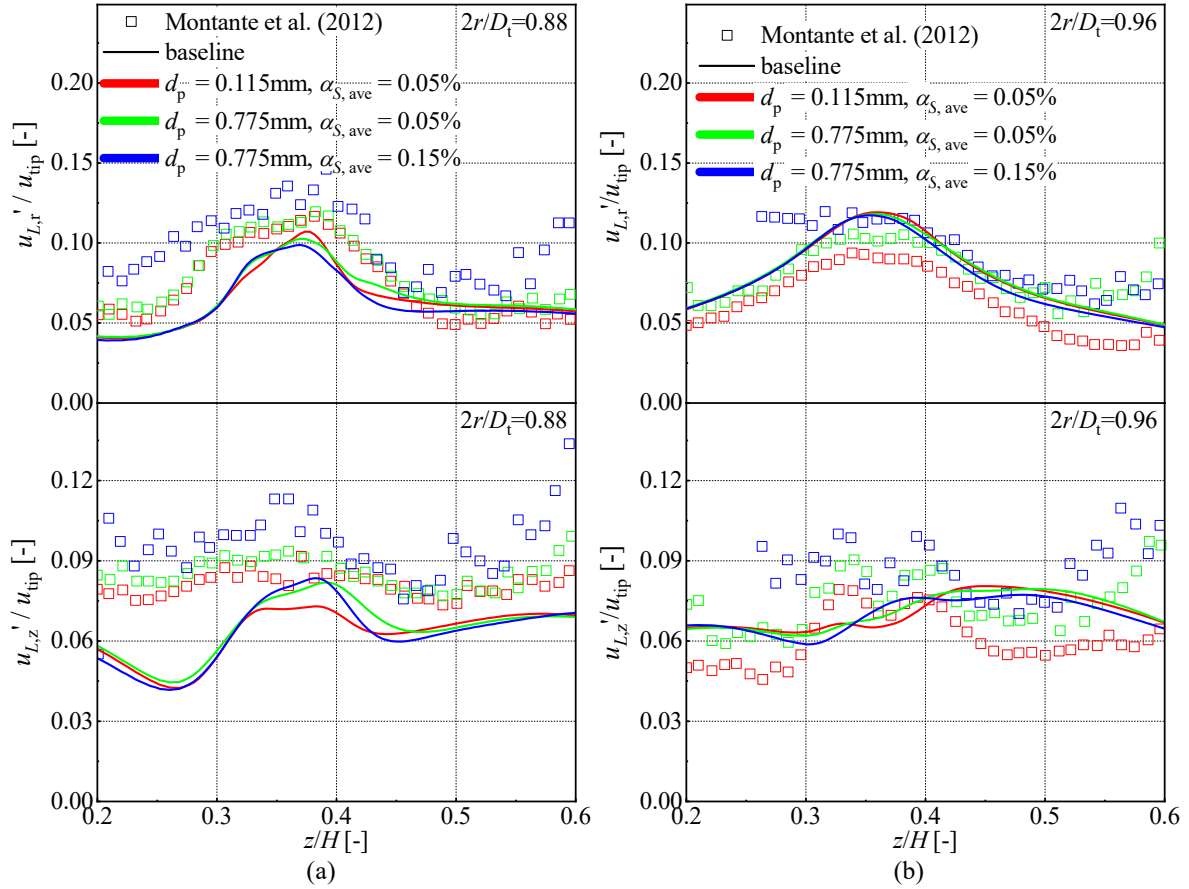
882 Figure 14 compares the baseline predictions for the axial profiles of the radial and axial mean liquid
 883 velocity to the experimental data from Montante et al. (2012) at the radial position of $2r/D_t =$
 884 0.88 . According to the experimental results, increasing the diameter of the suspended glass
 885 particles from 0.115 to 0.775 mm while keeping the average solids loading at $\alpha_{S,ave} = 0.05\%$
 886 apparently does not change the radial velocity component but tends to decrease the axial component
 887 in the height range above the impeller, i.e. for $0.3 \leq z/H \leq 0.6$. On the other hand, increasing the
 888 average solids loading from 0.05% to 0.15% while keeping the particle diameter of 0.775 mm
 889 results in a decrease in the magnitude of the axial component outside the impeller stream, namely

890 for $0.2 \leq z/H \leq 0.3$ and $0.45 \leq z/H \leq 0.6$. The agreement of the baseline predictions with the
 891 experimental data is overall acceptable with notable deviations seen in the impeller stream for the
 892 radial velocity and near the tank bottom and the liquid surface in the tangential velocity. Despite
 893 these significant absolute deviations, the corresponding predictions are able to represent most of
 894 the trends concerning variation of particle size and solid fraction.



895
 896 Figure 14. Comparison of the simulation results for the baseline model (lines) and measured data (symbols) from
 897 Montante et al. (2012) for the radial and axial components of mean liquid velocity. Axial profiles restricted to the
 898 height range of $0.2 \leq z/H \leq 0.6$ are shown at the radial position of $2r/D_t = 0.88$ and for different operation
 899 conditions (indicated by different colors) concerning particle size and solids loading.

900 A similar comparison concerning the fluctuating liquid velocity is shown in Figure 15. For this
 901 parameter, experimental results are provided for the radial and axial components at two radial
 902 positions $2r/D_t = 0.88$ and $2r/D_t = 0.96$. As seen from the experimental data, overall the
 903 magnitude of the radial and axial velocity components increases both with increasing mean solids
 904 loading and with increasing particle size. The baseline predictions agree well with the experimental
 905 data for the radial velocity fluctuations, while notable deviations are seen for the axial component.
 906 The predictions do not change too much between the three different operation conditions so that
 907 no clear dependency on particle size or solid fraction can be distinguished.

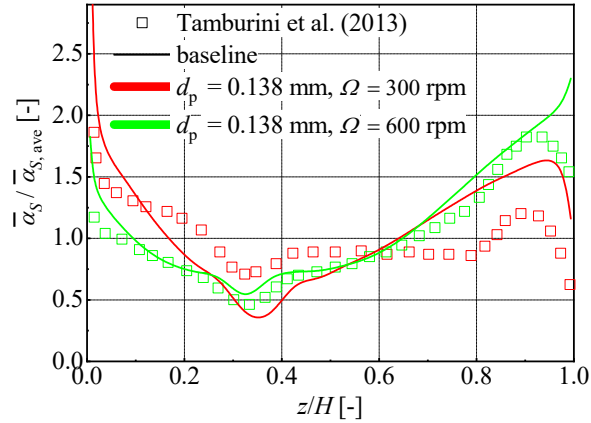


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909

910 Figure 15: Comparison of the simulation results for the baseline model (lines) and measured data (symbols) from
 911 Montante et al. (2012) for the radial and axial components of fluctuating liquid velocity. Axial profiles restricted to
 912 the height range of $0.2 \leq z/H \leq 0.6$ are shown for different operation conditions (indicated by different colors)
 913 concerning particle size and solids loading at the radial position of (a) $2r/D_t = 0.88$ and (b) $2r/D_t = 0.96$.

914 5.2.4 Tests from Tamburini et al. (2013)

915 The tests from Tamburini et al. (2013) provide data for the radially averaged axial profiles of solid
 916 fraction at the two impeller rotation speeds of 300 and 600 rpm at the same particle size and average
 917 solid fraction. As seen from Figure 16, the measured solid fraction for both values of Ω decreases
 918 starting from the tank bottom and reaches a minimum near the location of the impeller, i.e. at
 919 $z/H \approx 0.35$. For $\Omega = 300$ rpm the profile then is almost flat between $0.4 \leq z/H \leq 0.8$, while for
 920 $\Omega = 600$ rpm it increases steadily. For both values of Ω the solid fraction reaches a maximum
 921 around $z/H \approx 0.9$ and then decreases again towards the liquid surface. In the upper/lower part of
 922 the tank the solid fraction is higher for the higher/lower value of the rotation rate, with the crossover
 923 point located around $z/H \approx 0.6$. This is obviously due to the fact that at a higher impeller rotation
 924 speed, a larger amount of particles can be suspended into the upper part of the tank. These
 925 qualitative features above are well captured by the predictions except for the flat part of the profile
 926 at $\Omega = 300$ rpm. Quantitatively, the agreement is very good at $\Omega = 600$ rpm except close to the tank
 927 bottom and liquid surface, where predicted values are too high. At $\Omega = 300$ rpm the prediction
 928 suffers under- and overestimations in the impeller stream and in the region near the top wall,
 929 respectively. Overall the agreement is still reasonable.



930
 931 Figure 16. Comparison of the simulation results for the baseline model (lines) and measured data (symbols) from
 932 Tamburini et al. (2013) for the solid fraction. Radially averaged axial profiles over the entire tank height are shown.

933 6 SUMMARY AND CONCLUSIONS

934 This paper is devoted to the establishment of a two-fluid Euler-Euler model for solid-liquid flows
 935 in stirred tanks. Focus has been on the modeling of interfacial forces which include drag, lift, virtual
 936 mass, and turbulent dispersion. Based on a comprehensive review of existing results from
 937 analytical, numerical, and experimental studies a set of closure relations representing the best
 938 currently available description of each aspect has been proposed as a baseline model. Several other
 939 model variants that originate from different combinations of interfacial force correlations were
 940 considered to highlight the importance of various aspects. To validate the model, a data set
 941 comprising mean liquid and solid velocities, turbulent fluctuations and solid fraction measurements
 942 was assembled from different sources in the literature. In this way all aspects of the overall model
 943 could be assessed.

944 Single-phase test cases were considered first to provide a reference for the assessment of the two
 945 phase flow simulations. The SSG RSM turbulence model in conjunction with the mixing-plane
 946 MRF method were adopted. The comparisons together with those from Shi and Rzehak (2018) for
 947 both the mean and fluctuating velocities have shown that good predictions are obtained at lower
 948 rotation speeds Ω up to ≈ 200 rpm, while deviations occur at higher values certainly from 850 rpm
 949 on. In the latter case only qualitative features of the data are reproduced. Although reasonable
 950 agreement for engineering purposes in line with previous works (Murthy and Joshi 2008, Shi and
 951 Rzehak, 2018) was found, improvements to the SSG RSM clearly remain desirable, which is still
 952 the subject of ongoing research (Launder and Sandham, 2002; Morsbach, 2016).

953 On the basis of these findings, investigation of the two-phase test cases proceeded with the
 954 proposed baseline model and seven reduced model variants summarized in Table 8. In particular,
 955 the value of the constant as $C_T = 0.224$ determining the integral timescale T_L^L was verified from
 956 the axial and radial profiles of the solid fraction. In addition, the necessity to modify the drag
 957 correlation of Schiller and Naumann (1933) by a Stokes-number dependent factor, namely Eqs.
 958 (18) and (19) in the presently proposed model, could be deduced from these data as well as the
 959 need for a Stokes number dependence in the turbulent dispersion, which is contained in the PDF-
 960 based model of Reeks (1991) and de Bertodano (1998) but not in the FAD approach of Burns et
 961 al., 2004). Lift and virtual mass forces were found negligible in the present test cases. However,
 962 these findings are strongly interdependent on one another. For example with a previous drag

963 modification factor from Lane et al. (2005), the lift force did have a significant impact on the results.
964 Therefore, in general it is recommended to use a complete model, accounting for possible effects
965 of lift and virtual mass as well as turbulent dispersion and a modified drag force.

966 The capability of the baseline model in reproducing the fluid flow field as well as in describing the
967 change in solid fraction distribution due to the change in impeller rotation speed was then assessed.
968 Good agreement with the experimental data was obtained for the mean liquid velocity and the solid
969 fraction, while for the liquid velocity fluctuation the agreement was only mediocre. This deviation
970 originates partly from the SSG RSM turbulence model, from which even in the single-phase tests
971 the fluctuation were not captured very well. In addition, neglect of the turbulence modulation due
972 to the presence of the dispersed phase (PIT), for which advanced models are still in preparation
973 (Ma, 2017), may also contribute.

974 Concerning further model development, including a model for the PIT is clearly needed. The use
975 of DNS simulations like in the work of Ma (2017) appears most promising in this direction. There
976 the anisotropic nature of the PIT should be taken into account (Parekh and Rzehak, 2018; Ma et
977 al., 2020). In addition, the model for the modification of the drag force due to turbulence is still in
978 a preliminary stage. The validity of the presently proposed correlation, Eq. (19), in the range of
979 $St > 1$ is still uncertain. Further data, either from experiment or from DNS simulation, are needed
980 on this range. In addition, inclusion of the lengthscale ratio d_p/Λ as a third parameter is necessary
981 for a complete description.

982 The development of better models should be accompanied by the acquisition of more accurate and
983 more comprehensive data for validation. In particular the availability of mean liquid and solid
984 velocities, turbulent fluctuations and solid fractions for the same configuration would be very
985 beneficial to interpret the simulation results. Also parametric variations of particle size, density
986 ratio, mean solids loading, and impeller rotation speed are largely lacking. Finally, the investigation
987 of polydisperse flows would be highly relevant to technical applications.

988 7 ACKNOWLEDGEMENT

989 Mr. Pengyu Shi acknowledges support from the Chinese Scholarship Council (CSC).
990 Computational resources were provided by HZDR. We also thank Dr. Graeme Lane for valuable
991 discussions on drag modification, Professor Jos Derksen for advices on the estimation of ratio of
992 lift to drag forces, and Professors Vivek V. Ranade and Martin Lopez-De-Bertodano for valuable
993 discussions on turbulent dispersion.

994 8 APPENDIX A. GRID INDEPENDENCY STUDY

995 Four different grids are employed to ascertain the grid independence as detailed in Table 9. The
996 cumulative distribution of the comprehensive parameters of mesh quality, i.e. the equiangular
997 skewness, the smoothness (maximum ratio of the volume of a cell to that of each neighboring cell),
998 and the aspect ratio (length ratio of the longest edge to the shortest edge), of the finally used mesh
999 3 are plotted in Figure 17. Distributions found for the other meshes behave similarly. The maximum
1000 values of these three parameters are roughly 0.5, 5, and 1.35, indicating a good mesh quality.

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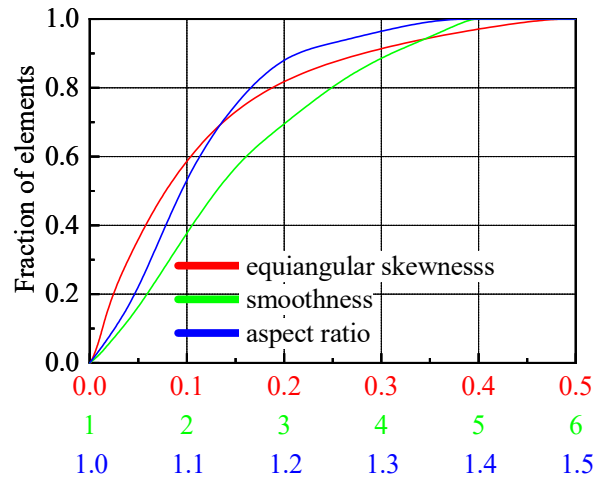
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Table 9: Parameters for meshes used in grid independency study.

Mesh	Tank volume			Impeller blade			Overall	CPU time (with 32 processors)
	N_r	N_θ	N_z	N_r	N_θ	N_z	N_{tot}	
1	101	72	106	20	2	16	7.7×10^5	64 h
2	123	90	130	24	3	25	1.44×10^6	120 h
3	101	120	120	30	4	30	1.45×10^6	130 h
4	160	120	150	30	4	30	2.88×10^6	300 h

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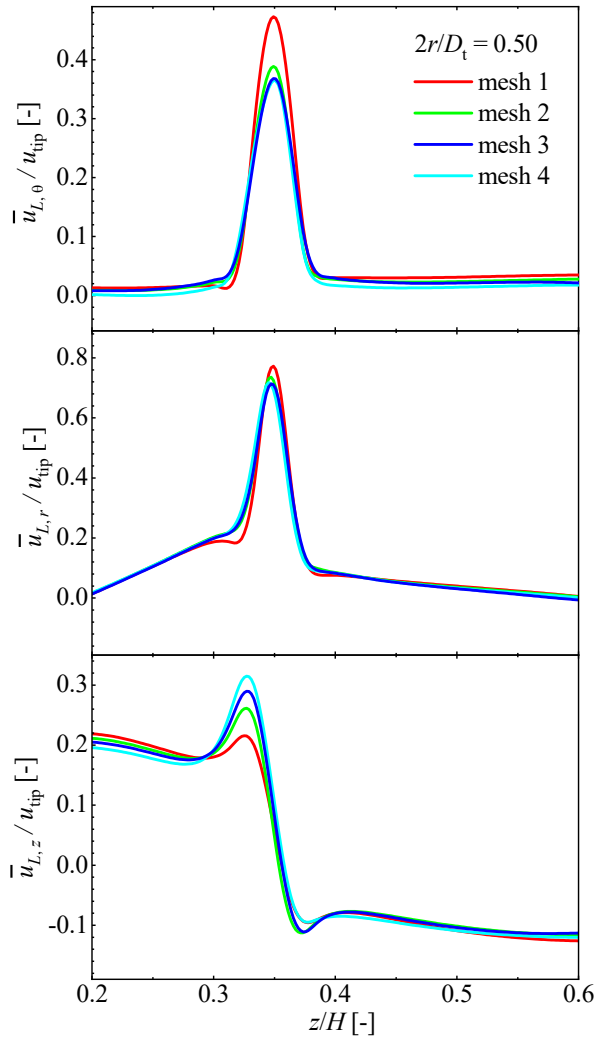
Figure 17. Cumulative distribution of the three measures of mesh quality, namely the equiangular skewness, the smoothness, and the aspect ratio (represented by the red, green, and blue lines, respectively), of mesh 3 as listed in Table 9.

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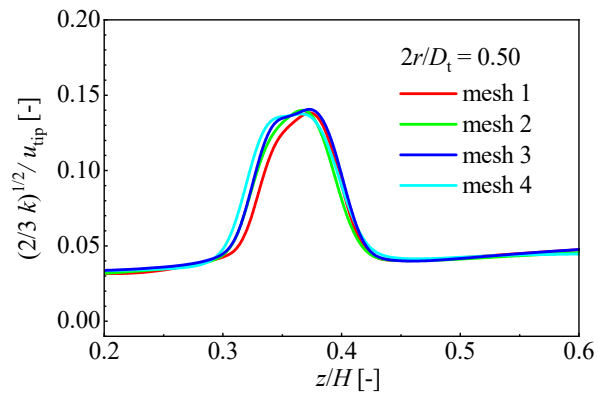
To illustrate the influence of the grid the test case of Guha et al. (2007) is presented, which is most critical due to the high impeller rotation rate (see Table 6 for details of experimental parameters). Results are shown for the axial profiles of mean and fluctuation velocities at $2r/D_t = 0.50$ in the plane mid-way between two baffles. The numerical settings described in section 4.2 are applied.

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Figure 18 shows the computational profiles of the three mean velocity components – tangential, radial, and axial – for each mesh. It is seen that the prediction of tangential and axial velocities within the impeller stream is significantly affected by the grid resolution. Results for meshes 3 and 4 show quite good agreement with each other, suggesting that the grid independence has been achieved for mesh 3. A similar comparison concerning the modeled fluctuation velocity $\sqrt{2/3} k$ is shown in Figure 19. The difference between the predictions according to meshes 2, 3, and 4 are vanishingly small, indicating a negligible influence of grid resolution here. These observations are consistent with those made in our previous investigation (Shi and Rzehak, 2018). In view of the computational time listed in the last column of Table 9, mesh 3 can be considered to give satisfactory results and meshes with similar average spacings in radial, azimuthal, and axial direction are generated for the other investigated cases (see Table 7).



1026
 1027 Figure 18. Results of grid independency study for the tangential (top panel), radial (middle panel), and axial (bottom
 1028 panel) components of mean liquid velocity. The case considered here is the single phase flow in Guha et al. (2007)
 1029 with an impeller rotation speed of 1000 rpm. Axial profiles restricted to a height range around the impeller are shown
 1030 at the radial position of $2r/D_t = 0.50$.



1031
 1032 Figure 19. Same as Figure 18 but for the fluctuation velocity.

1033

1034 9 NOMENCLATURE

1035

Notation	Unit	Denomination
Latin formula characters		
A_{ij}, \mathbf{A}	-	anisotropy tensor
C_D	-	drag coefficient
$C_{D,0}$	-	stagnant drag coefficient
$C_{D,T}$	-	turbulent drag coefficient
C_i	m	clearance between the turbine and tank bottom
C_L	-	lift coefficient
$C_{L\omega}$	-	shear-induced lift coefficient
$C_{L\Omega}$	-	spin-induced lift coefficient
C_T	-	constant in describing T_L^L
C_{VM}	-	virtual mass force coefficient
C_Λ	-	constant in describing Λ
d_p	m	particle diameter
D_{dis}	m	disk diameter
D_i	m	impeller diameter
D_t	m	tank diameter
D_{ij}, \mathbf{D}	s ⁻¹	strain rate tensor
\mathbf{F}	N m ⁻³	force per unit volume
g	m s ⁻²	acceleration of gravity
H	m	tank filled height
H_{bla}	m	blade height
\mathbf{I}	-	identity tensor
$J(\epsilon)$	-	function defined by McLaughlin (1991) Eq. (20)
k	m ² s ⁻²	turbulent kinetic energy
N	-	number of grid cells
p	Pa	pressure (static)
r	m	radial coordinate
R_{ij}, \mathbf{R}	m ² s ⁻²	Reynolds stress tensor
$Re_p = u_{rel}d_p/\nu$	-	Reynolds number based on relative velocity
$Re_\omega = \omega d_p^2/\nu$	-	Reynolds number based on flow vorticity
$Re_\Omega = \Omega d_p^2/\nu$	-	Reynolds number based on particle rotation rate
$Rr = \Omega d_p/u_{rel}$	-	dimensionless particle rotation rate

$Sr = \omega d_p / u_{rel}$	-	dimensionless flow vorticity or shear rate
St	-	Stokes number
t	s	time
\mathbf{T}	N m^{-2}	stress tensor
T_L^L	s	Lagrangian integral timescale following the fluid motion
T_L^S	s	Lagrangian integral timescale following the particle motion
u, \mathbf{u}	m s^{-1}	resolved velocity
\mathbf{u}'	m s^{-1}	fluctuating velocity
$\bar{\mathbf{u}}$	m s^{-1}	averaged velocity
u_{rel}	-	slip velocity
$u_{term,0}$	m s^{-1}	stagnant terminal velocity
$u_{term,T}$	m s^{-1}	turbulent terminal velocity
u_{tip}	m s^{-1}	impeller tip velocity
W_{baf}	m	baffle width
W_{bla}	m	blade width
W_{ij}, \mathbf{W}	s^{-1}	rotation rate tensor
y	m	wall normal coordinate
z	m	axial coordinate with the origin at the tank bottom
z_{bla}	m	axial coordinate with the origin at the impeller disk
Greek Formula characters		
$\bar{\alpha}$	-	phase fraction
β	-	turbulence structure parameter
δ_{ij}	-	Kronecker delta
$\epsilon = \sqrt{Sr/Re_p}$	-	dimensionless length ratio
ϵ	$\text{m}^2 \text{s}^{-3}$	turbulent dissipation rate
Λ	m	Eulerian longitudinal integral lengthscale
μ	$\text{kg m}^{-1} \text{s}^{-1}$	dynamic viscosity
ν	$\text{m}^2 \text{s}^{-1}$	kinematic viscosity
θ	rad	azimuthal angle
ρ	kg m^{-3}	density
τ_{cross}	s	time for a particle to cross an typical eddy
τ_S	s	particle relaxation time
ω	s^{-1}	flow vorticity
Ω	rpm.	impeller rotation speed
Ω_{fr}, Ω_{fr}	s^{-1}	particle angular velocity / rotation rate in the torque-free condition
Latin indices		

body	-	on body
k	-	k^{th} phase
i, j	-	cartesian vector / tensor components
inter	-	on interface
L	-	liquid phase
mol	-	molecular
S	-	solid phase
turb	-	turbulent

1036

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